

X.5

FINAL ORAL REPORT

**ORBITING RESEARCH LABORATORY /
LOGISTICS SPACECRAFT CHECKOUT
REQUIREMENTS STUDY**

MARCH 1965

TRW SPACE TECHNOLOGY LABORATORIES

THOMPSON RAMO WOOLDRIDGE INC.

FLORIDA DIVISION • 7001 NORTH ATLANTIC AVENUE • CAPE CANAVERAL, FLORIDA

INTRODUCTION

This report presents the results of the study of Orbiting Research Laboratory and Logistic Spacecraft Checkout Requirements as they relate to pre-launch operations at MILA. The study was performed jointly by the Florida Division and the Systems Research and Analysis Division of TRW Space Technology Laboratories for NASA under the terms of contract NAS10-1706.

The principal objectives of this study were to establish recommended modes of operation for the assembly, checkout and launch of specified Orbiting Research Laboratories and Logistics Spacecraft at MILA and to determine the resulting impact on KSC resources in terms of facilities, equipment, procedures and personnel.

The study encompassed all planned MILA pre-launch operations relating to Orbiting Research Laboratories, Logistic Spacecraft, crew and cargo. A summary of the study tasks and subtasks is presented in an accompanying chart. Particular emphasis was placed on the subtasks relating to facilities, acceptance checkout equipment (ACE) and scientific equipment (ORL experiments), since new requirements in these areas represented the greatest potential impact on MILA resources.

Various ORL, Logistics Spacecraft, and launch vehicle combinations were studied. The ORL's fall into three basic classes: The Apollo Extensions, the Single Module Space Station (SMSS), and the Multi-Module Space Station (MMSS). Three logistic spacecraft were studied: The Apollo LS, the Modified Apollo LS (MODAP), and the Ballistic LS (BALLOS). Launch vehicle study was confined to the Saturn IB and the Saturn V, and was restricted to spacecraft interface considerations.

LIST OF ABBREVIATIONS

ACE	Acceptance Checkout Equipment	MODAP	Modified Apollo
AE	Apollo Extensions	MORL	Manned Orbital Research Lab (DAC)
A/S	Ascent Stage	MOSS	Manned Orbital Space Station
BPC	Boost Protective Cover	NE	Nozzle Extension
BU	Backup	NS	Nozzle Simulator
CGM	Cargo Module	O&C	Operations & Checkout (Facility)
CM	Command Module	OFTL	Ordnance Field Test Lab
CSM	Command & Service Module (Mated)	ORL	Orbital Research Laboratory
D/S	Descent Stage	OSF	Ordnance Storage Facility
ECS	Environmental Control System	PIB	Pyrotechnic Installation Bldg
EDS	Emergency Detection System	RCS	Reaction Control System
EIF	Experiment Integration Function	RF	Radar-Boresight Facility
EIT	Experiment Integration & Test	S/C	Spacecraft
ELS	Earth Landing System	SCS	Stabilization Control System
FHS	Forward Heat Shield	SLA	Spacecraft LEM Adapter
G&N	Guidance and Navigation	SM	Service Module
ICA	Integrated Checkout Area	SMSS	Single-Module Space Station
LC	Launch Complex	SPS	Service Propulsion System
LEM	Lunar Excursion Module	STF	Static Test Facility
LES	Launch Escape System	S/V	Space Vehicle (L/V + S/C)
LET	Launch Escape Tower	SVP	Service Pack
LM(D)	Laboratory Module (Dependent)	S-IB	Saturn I-B
LM(I)	Laboratory Module (Independent)	S-V	Saturn V
LS	Logistics Spacecraft	TVA	Thrust Vector Alignment
L/V	Launch Vehicle	W&B	Weight and Balance
MMSS	Multi-Module Space Station		

STUDY OBJECTIVES

1. DETERMINE THE EFFECT OF ORL/LS ON MILA OPERATIONS
2. PROVIDE METHOD FOR EVALUATING FUTURE REQUIREMENTS

STUDY GROUND RULES

1. ORL/LS-NO INTERFERENCE WITH LUNAR APOLLO
2. MAXIMUM USE OF EXISTING FACILITIES AND EQUIPMENT
3. PRELAUNCH OPERATIONS ONLY (EXCEPT REFURBISHMENT & RESUPPLY)
4. ORL/LS & LUNAR APOLLO ONLY
5. SPACECRAFT & RELATED ACTIVITIES ONLY

LAUNCH SCHEDULE

The potential manned space flight schedule, revised 28 December 1964, was issued by NASA Headquarters for a study baseline. To facilitate analysis of the impact of this program on MILA resources, certain representative time periods have been selected from the baseline schedule; during these periods, significant step function changes occur in ORL activities.

Case I represents a change in launch rate of Apollo-type spacecraft, from three-month to two-month intervals. Also, the AE configuration (Apollo X-5) is introduced into the program during this period.

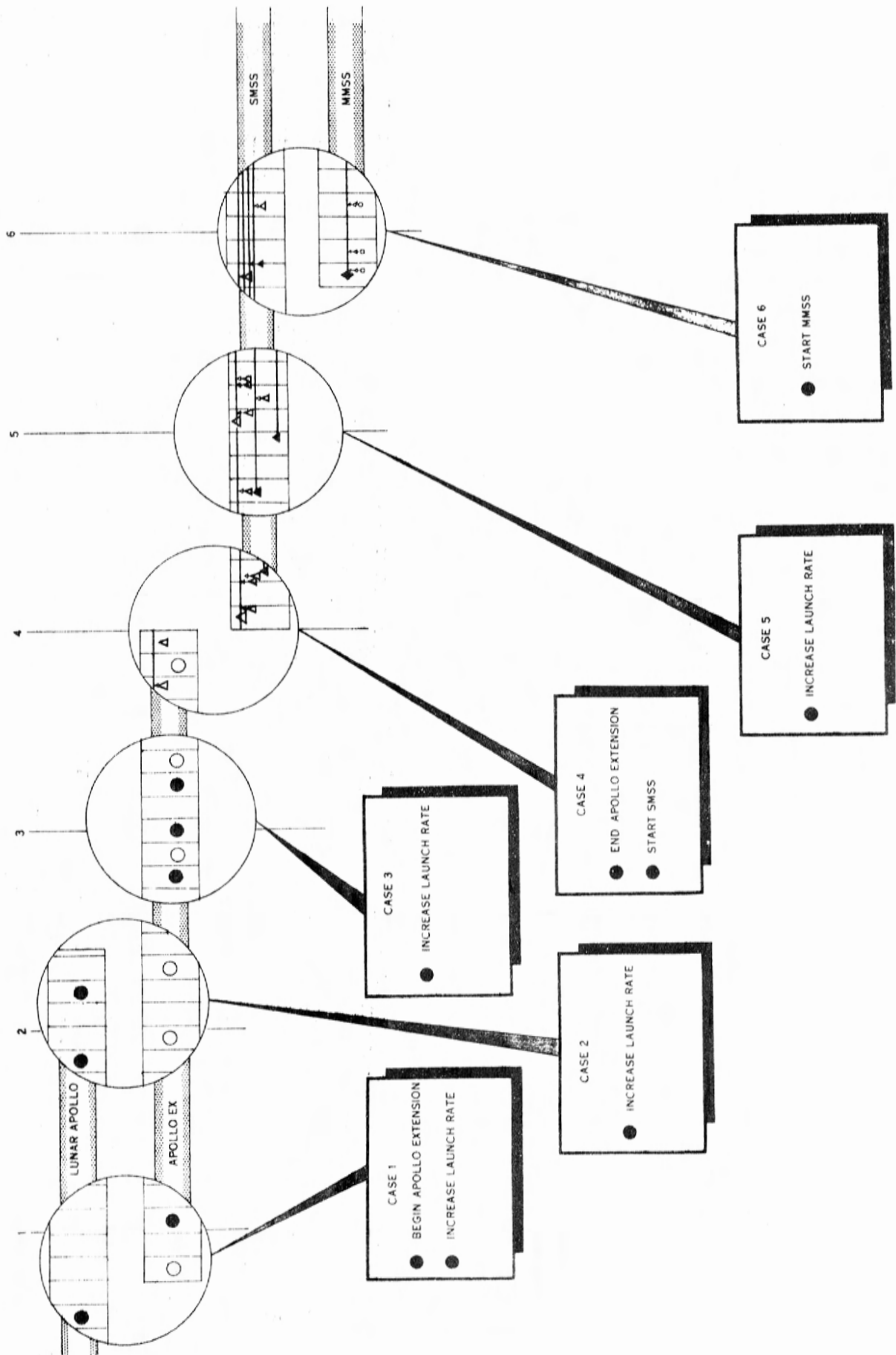
Case II shows two Apollo-type spacecraft (Lunar Apollo and X-6) launched at one-month intervals.

In Case III, three Apollo-type spacecraft (Lunar Apollo and X-6) are launched at one-month intervals.

In Case IV, the SMSS and the three-six man Logistic Spacecraft are introduced into the program, concurrently with the AE (Apollo X-6). AE resupply is also shown. Backup logistic spacecraft are shown as required and the SMSS/LS launch sequence is in accordance with the DAC MORL study. The SMSS launched by a Saturn V is manned at launch, with an Apollo Command Module installed in the hangar area.

Case V shows three SMSS's launched in a four-month period. Two of these are manned launches, utilizing a Saturn V. The third is a Saturn IB unmanned launch, which is then manned and supported in accordance with the DAC MORL Study. Logistic spacecraft with appropriate backup are indicated.

During the time period represented by Case VI, four SMSS's are in orbit, resupplied by the three-six man LS. The Multi-Module Space Station is launched and initial manning is accomplished by the DALLOS. Backup logistic spacecraft are provided in all cases.



SELECTION OF CASES FROM PROGRAM SCHEDULE

SELECTED CASES
FROM
PROGRAM LAUNCH SCHEDULE

SPACECRAFT	CALENDAR MONTHS								
	1	2	3	4	5	6	7	8	9
LUNAR APOLLO APOLLO EX. (X-5)	▲		△		▲		▲		

CASE I

SPACECRAFT	CALENDAR MONTHS								
	1	2	3	4	5	6	7	8	9
LUNAR APOLLO APOLLO EX. (X-6)	▲	△		▲	△				

CASE II

SPACECRAFT	CALENDAR MONTHS								
	1	2	3	4	5	6	7	8	9
AES (LUNAR) APOLLO EX. (X-6)	▲	△	▲		▲	△		△	▲

CASE III

SPACECRAFT	CALENDAR MONTHS								
	1	2	3	4	5	6	7	8	9
APOLLO EX. (X-6) 3-6 MAN LS SMSS	△		△ ●	▲	△	△	△	△ ●	▲

CASE IV

SPACECRAFT	CALENDAR MONTHS								
	1	2	3	4	5	6	7	8	9
3-6 MAN LS SMSS	△		▲	△	△	△			△

CASE V

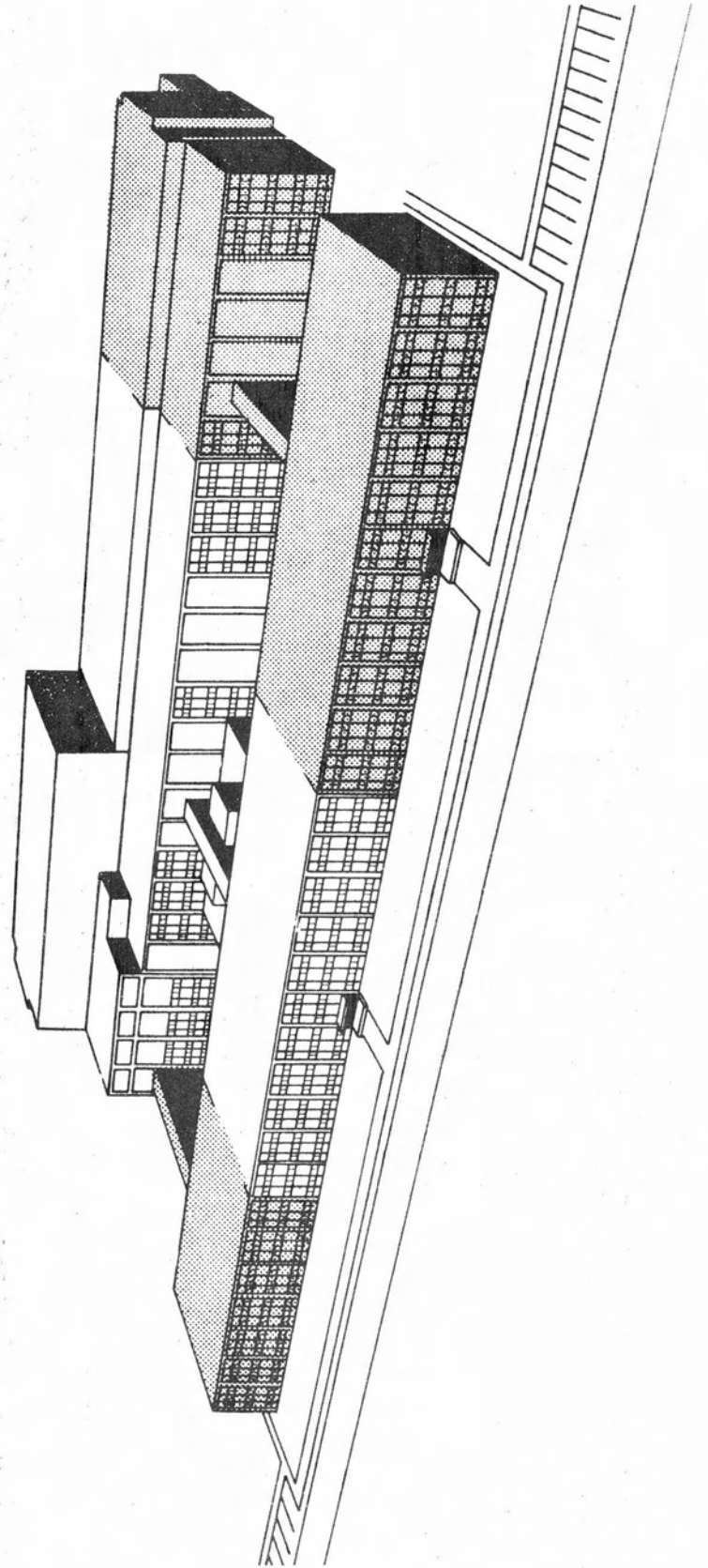
SPACECRAFT	CALENDAR MONTHS								
	1	2	3	4	5	6	7	8	9
3-6 MAN LS 3-6 MAN LS MMSS BALLOS	△		▲		△	▲		▲	△

CASE VI

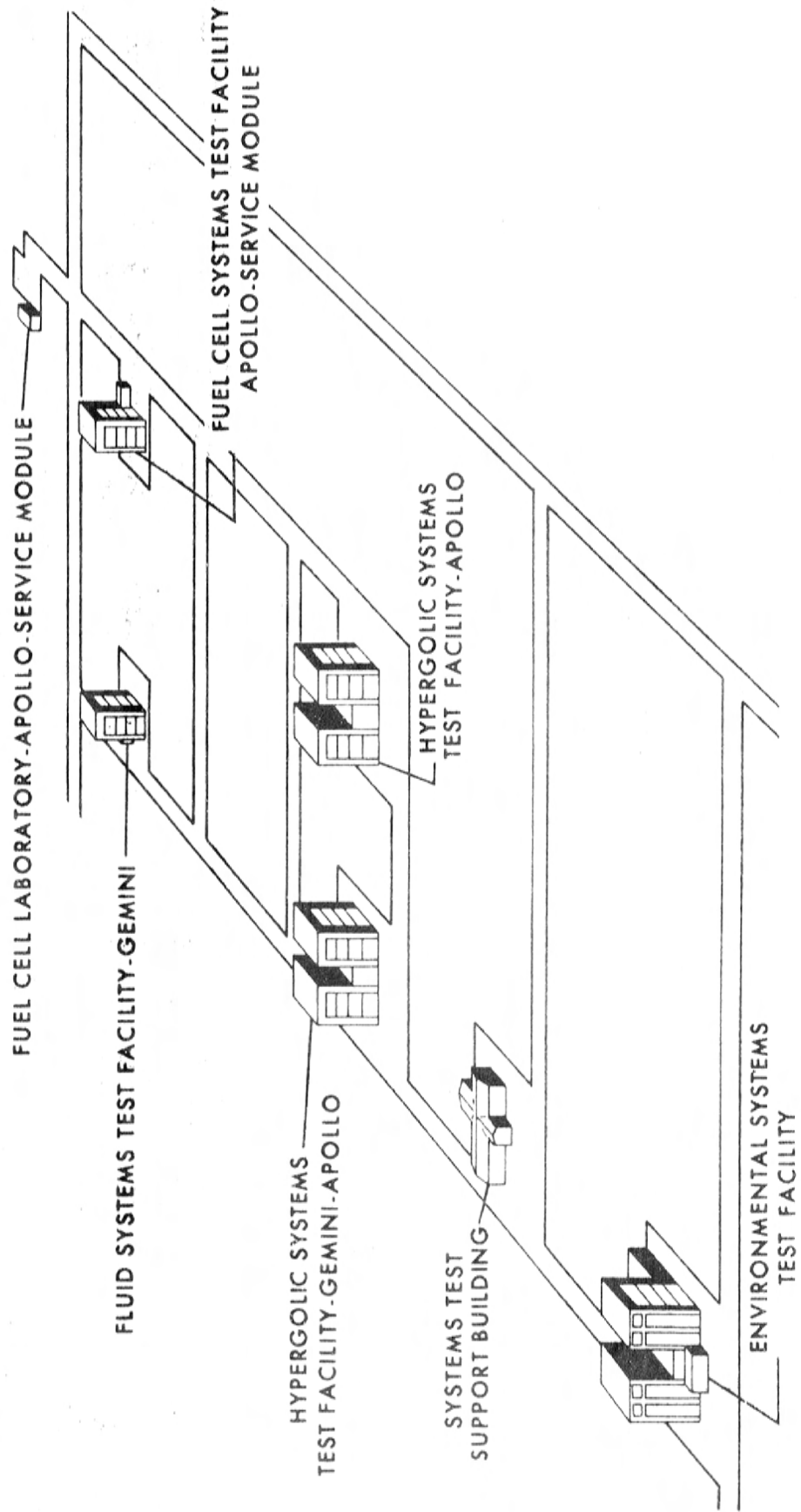
- △ S-IB LAUNCH
- ▲ S-V LAUNCH
- BACKUP LS READY
- BACKUP/EMERGENCY LS ON PAD

OPERATIONS AND CHECKOUT

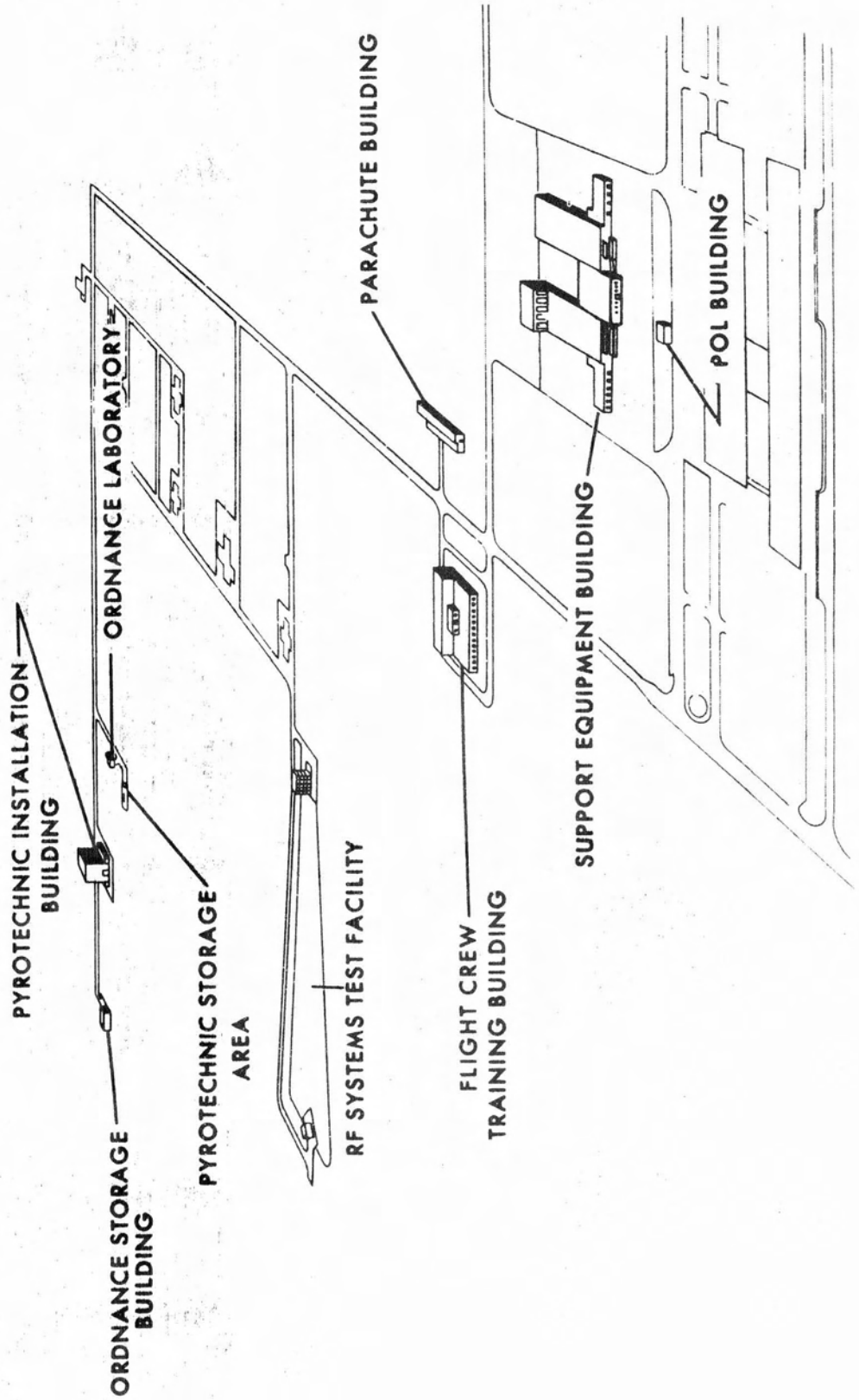
BUILDING



FLUID TEST COMPLEX



OTHER FACILITIES



STUDY CONFIGURATIONS

Apollo Extensions

Several configurations of the Apollo Extensions were selected as study models, to be used in the development of recommended modes of operation and for impact analysis.

The minimum mod Apollo was selected as the first study model and was designated Apollo X-1. A two-man crew was assumed, since this possible configuration represented a significant change from Lunar Apollo.

Both a dependent LEM shell and a dependent new Laboratory Module were considered for the second study model, designated Apollo X-2. The dependent Laboratory Module was selected, since it represented the major change from Lunar Apollo, and since greater impact on KSC resources was indicated.

The independent LEM laboratory was considered in a subsequent model, designated Apollo X-5. Ascent stage propulsion was assumed for this model, to provide a worst case basis for facilities and checkout equipment utilization.

The independent LEM laboratory with ascent stage propulsion removed combined with a modified LEM descent stage was established as another study model, designated Apollo X-6.

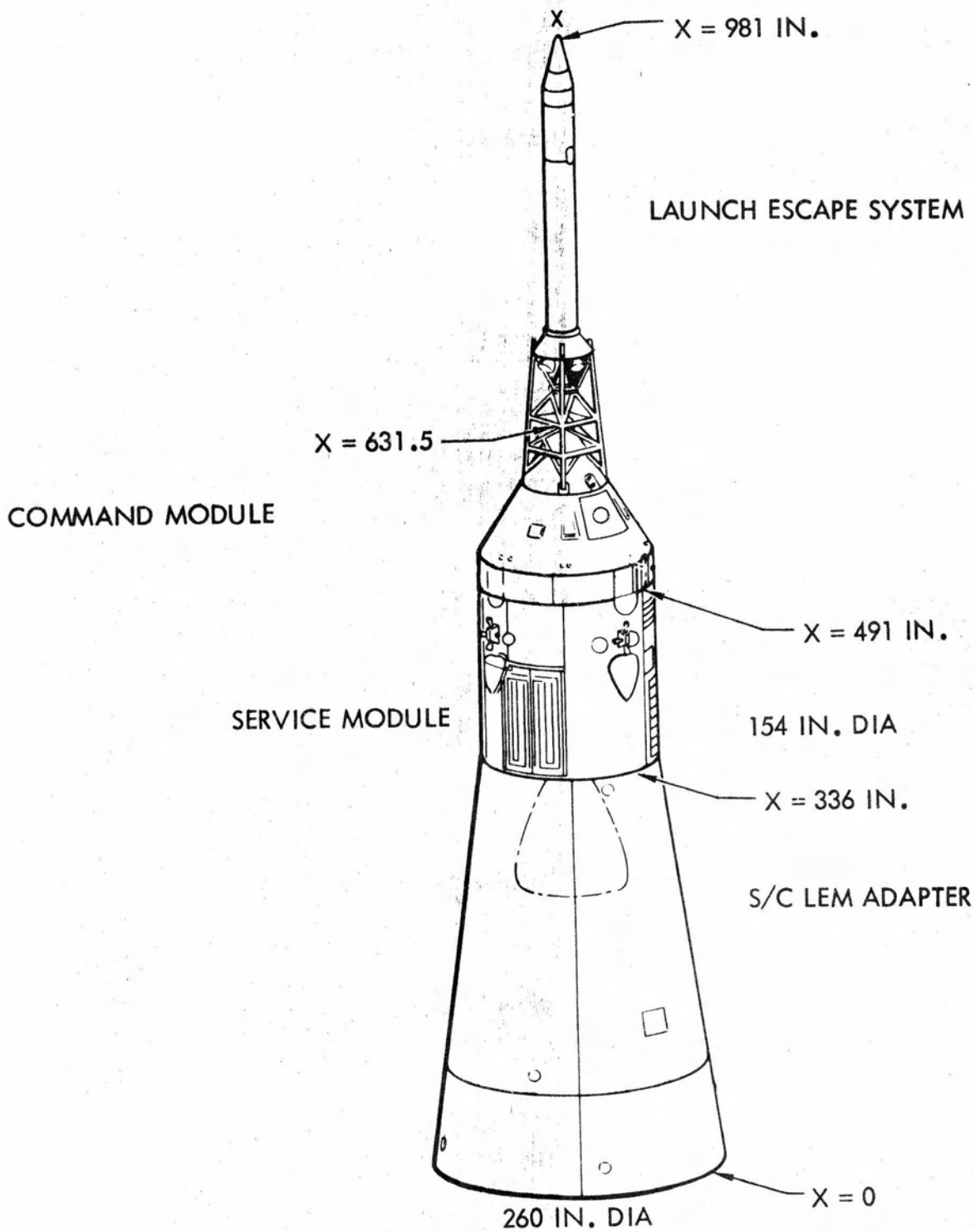
To complete the study models for Apollo Extensions, an Apollo CSM with both a dependent and independent Laboratory Module was selected. This was designated Apollo X-4.

Single Module Space Station (SMSS)

The Langley Research Center/DAC Manned Orbital Research Laboratory (MORL) was selected as a study model for this class space station. Both the Apollo and the Modified Apollo (MODAP) Logistic Spacecraft were considered in connection with the SMSS.

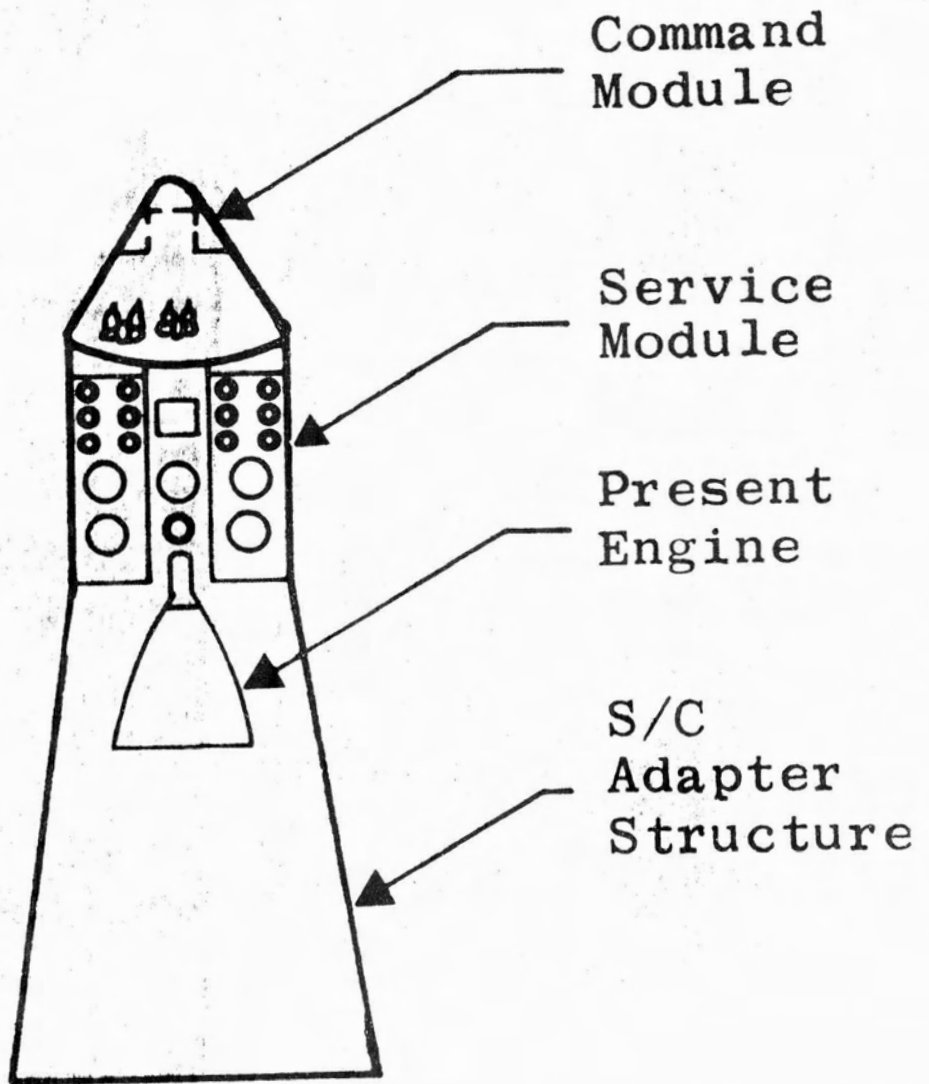
Multi-Module Space Station (MMSS)

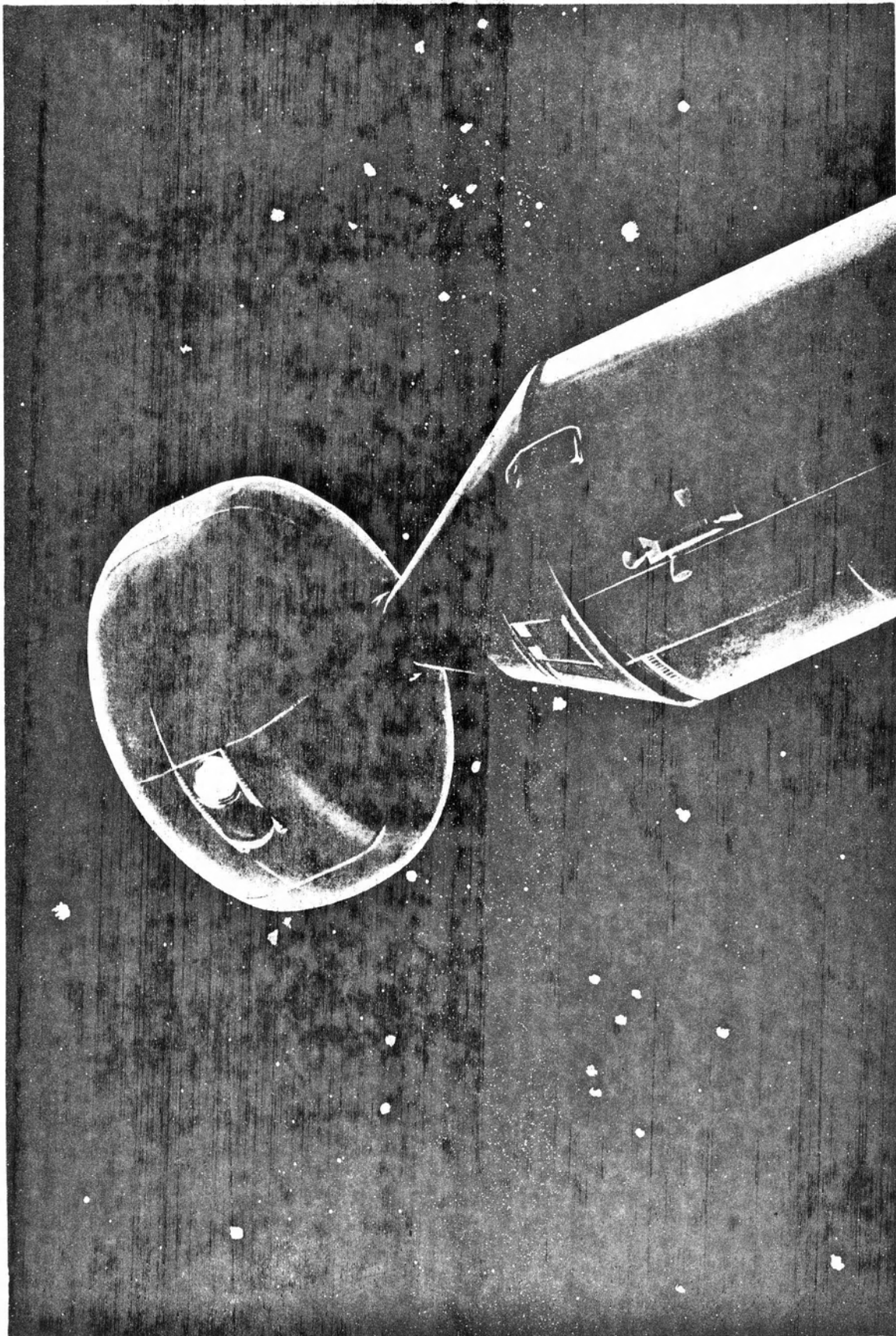
The study model selected was based on the Lockheed Study: three-arm radial segmented elements surrounding a central hub. This ORL is manned and supported by the Ballistic Logistic Spacecraft (BALLOS).



Apollo X-1 Launch Configuration

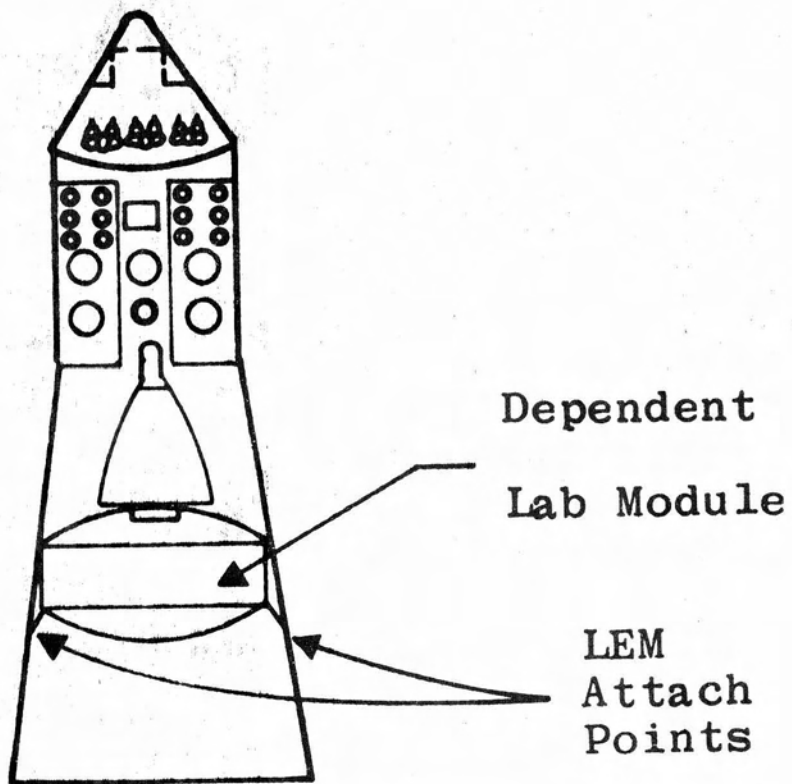
X-1
(2 man)



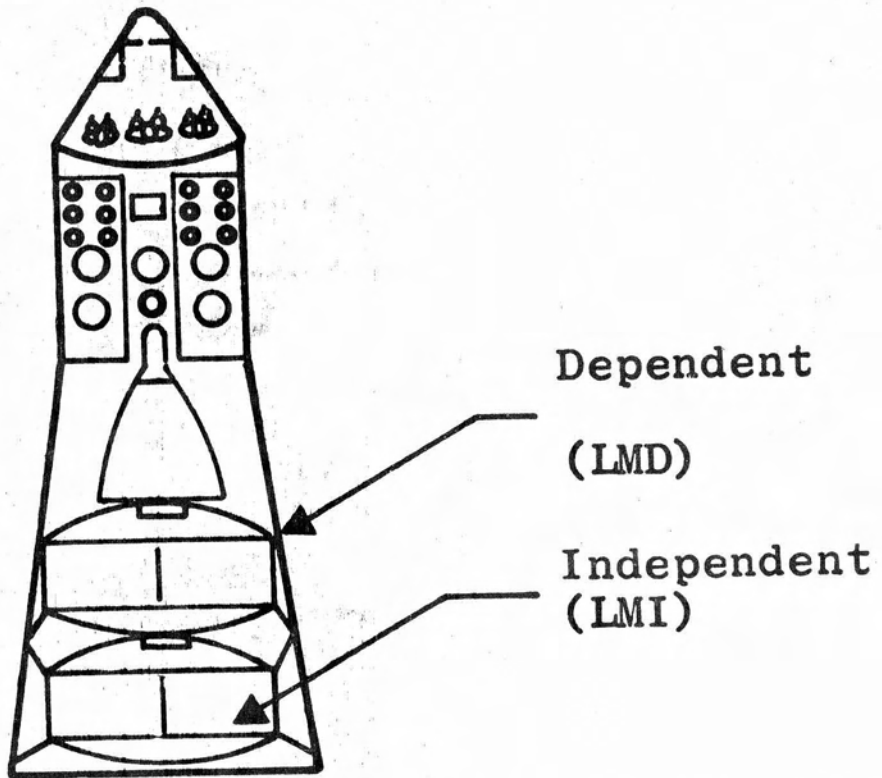


Apollo X-2 Configuration—Modified Command Module
and Dependent Laboratory Module

X-2
(3 man)



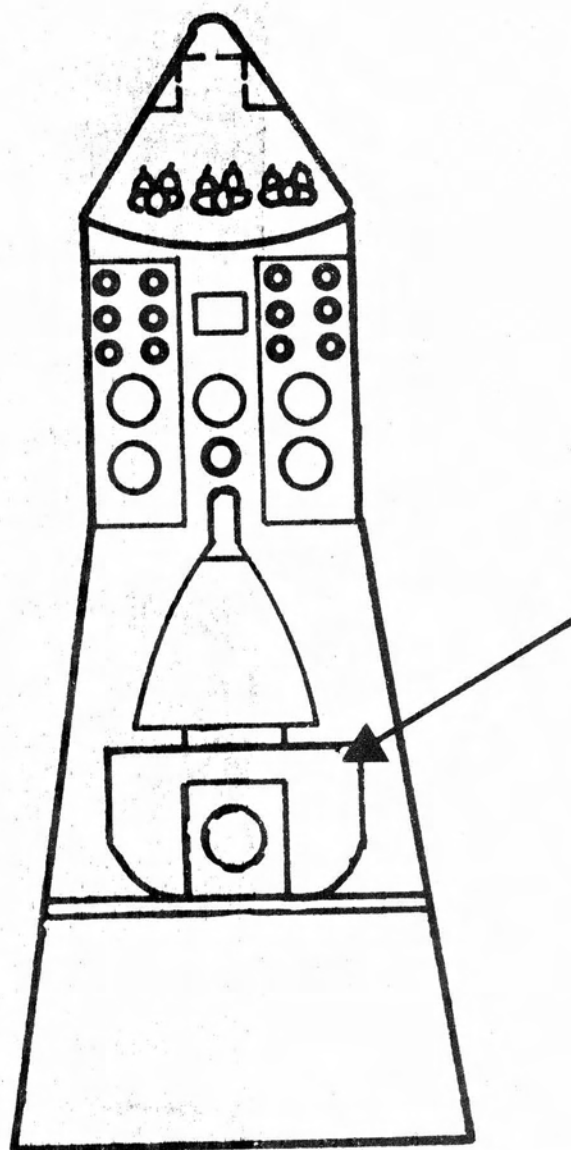
X-4
(3 man)





Apollo X-5 Configuration—Modified Command Module
and Modified LEM Ascent Stage

X-5
(3 man)



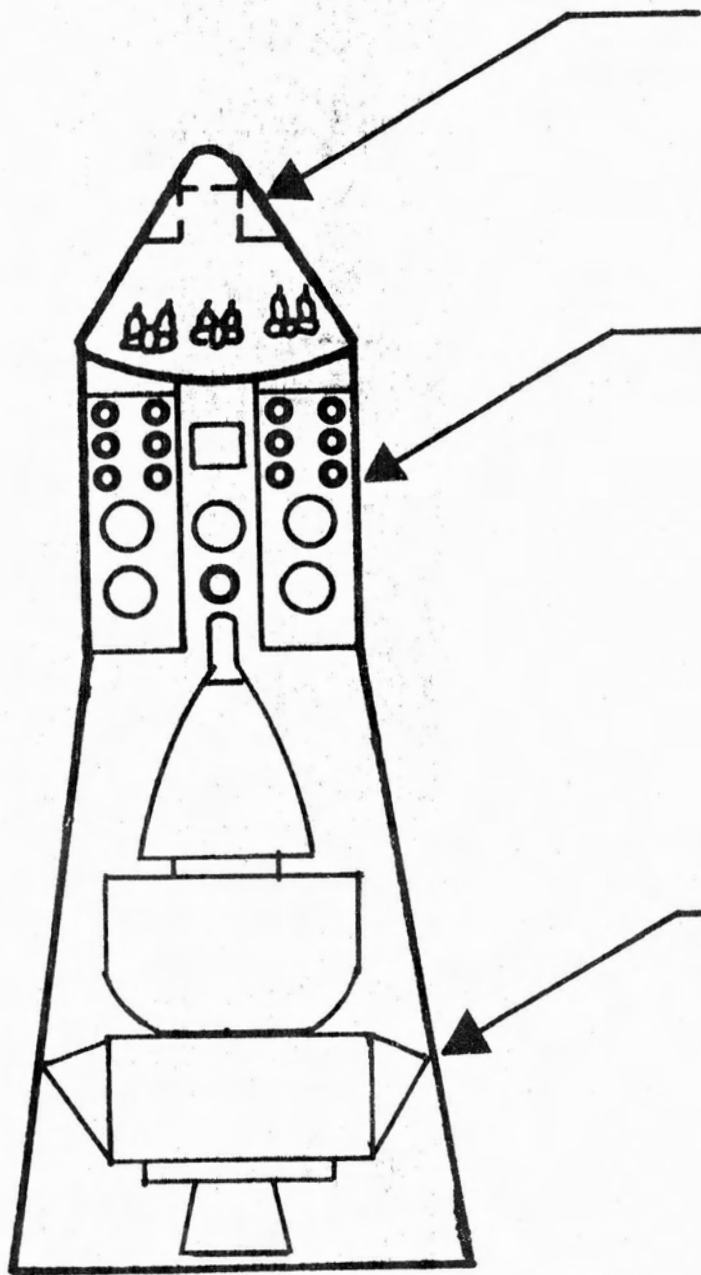
Modified
LEM
Ascent
Stage

X-6
(3 man)

Command
Module

Service
Module

S/C
Adapter
Structure

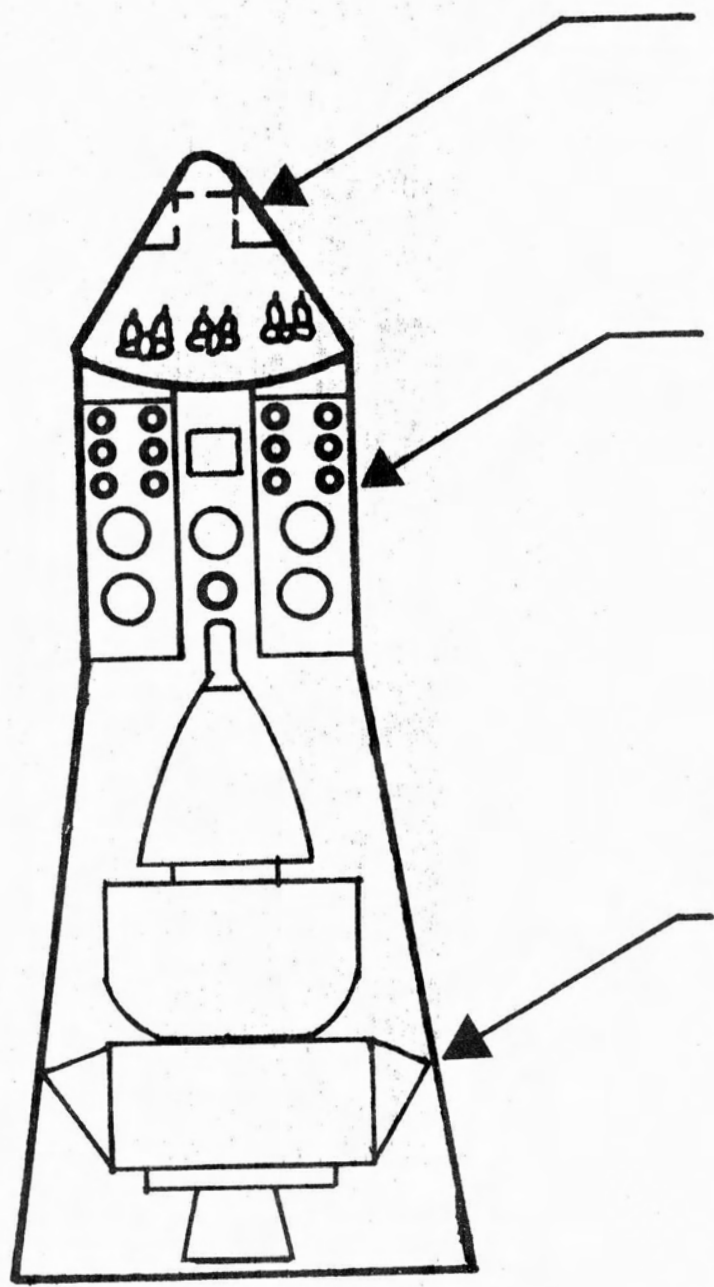


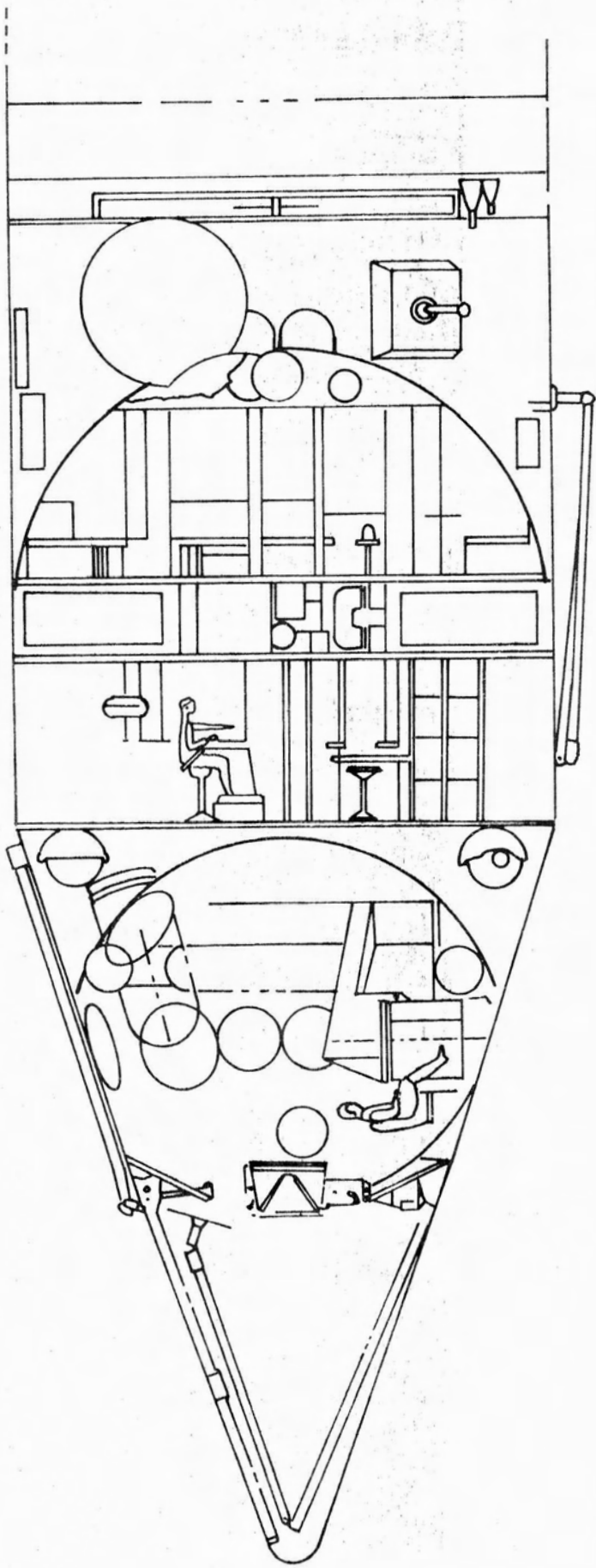
X-6
(3 man)

Command
Module

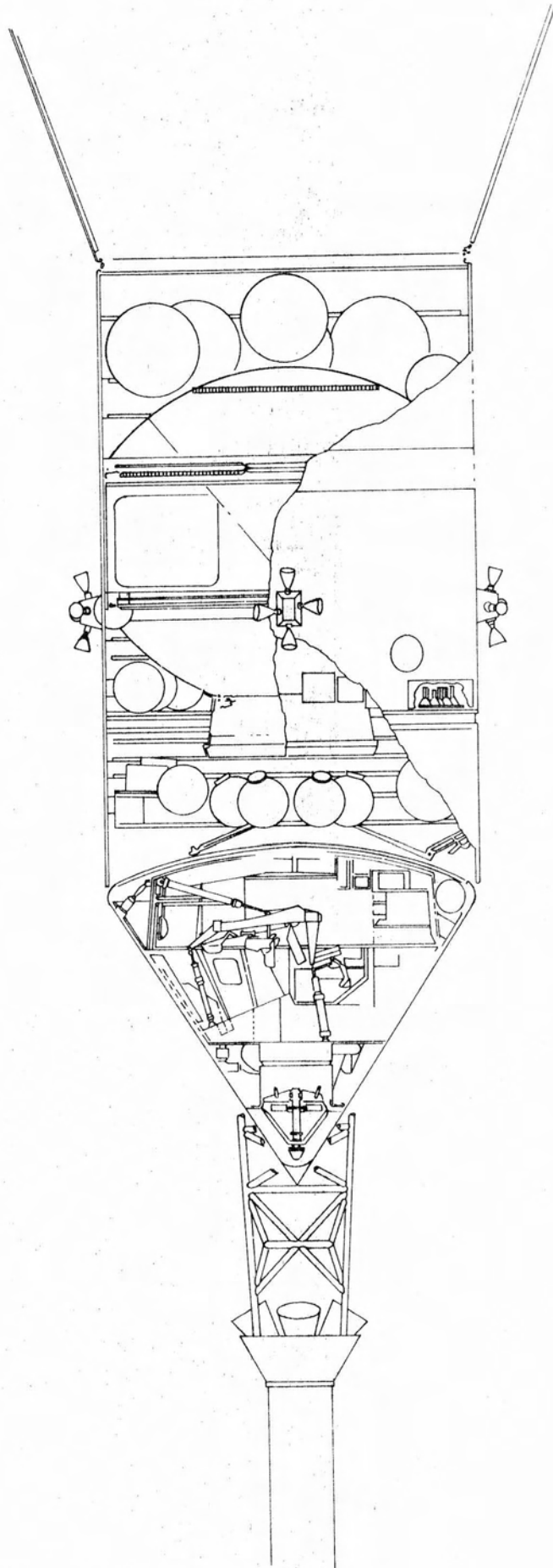
Service
Module

S/C
Adapter
Structure

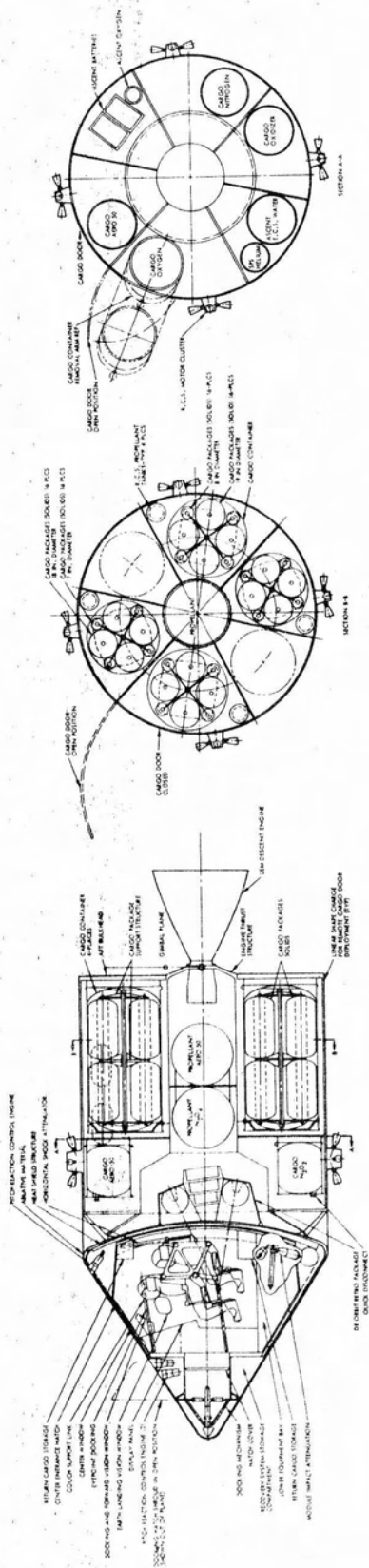




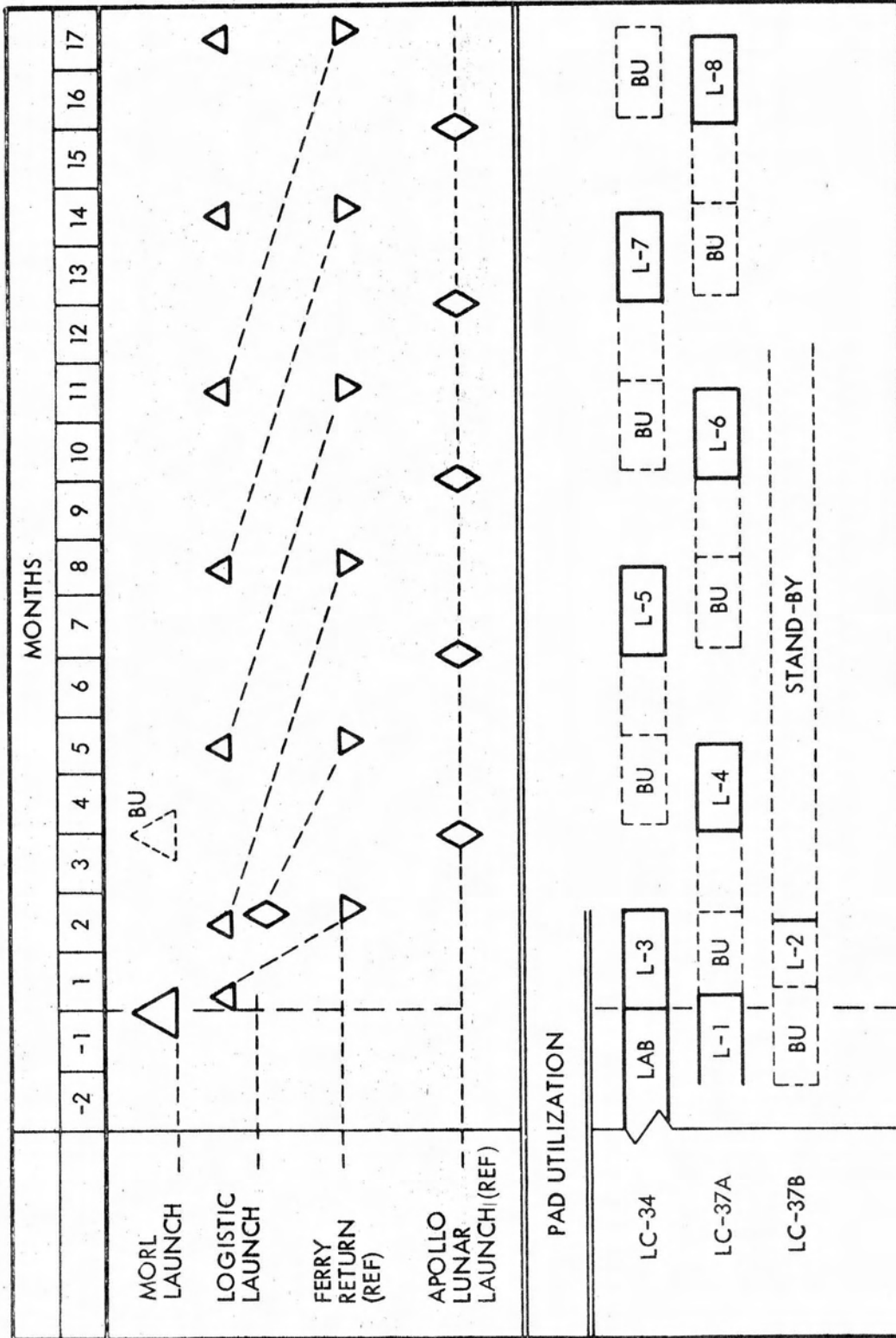
SINGLE MODULE SPACE STATION



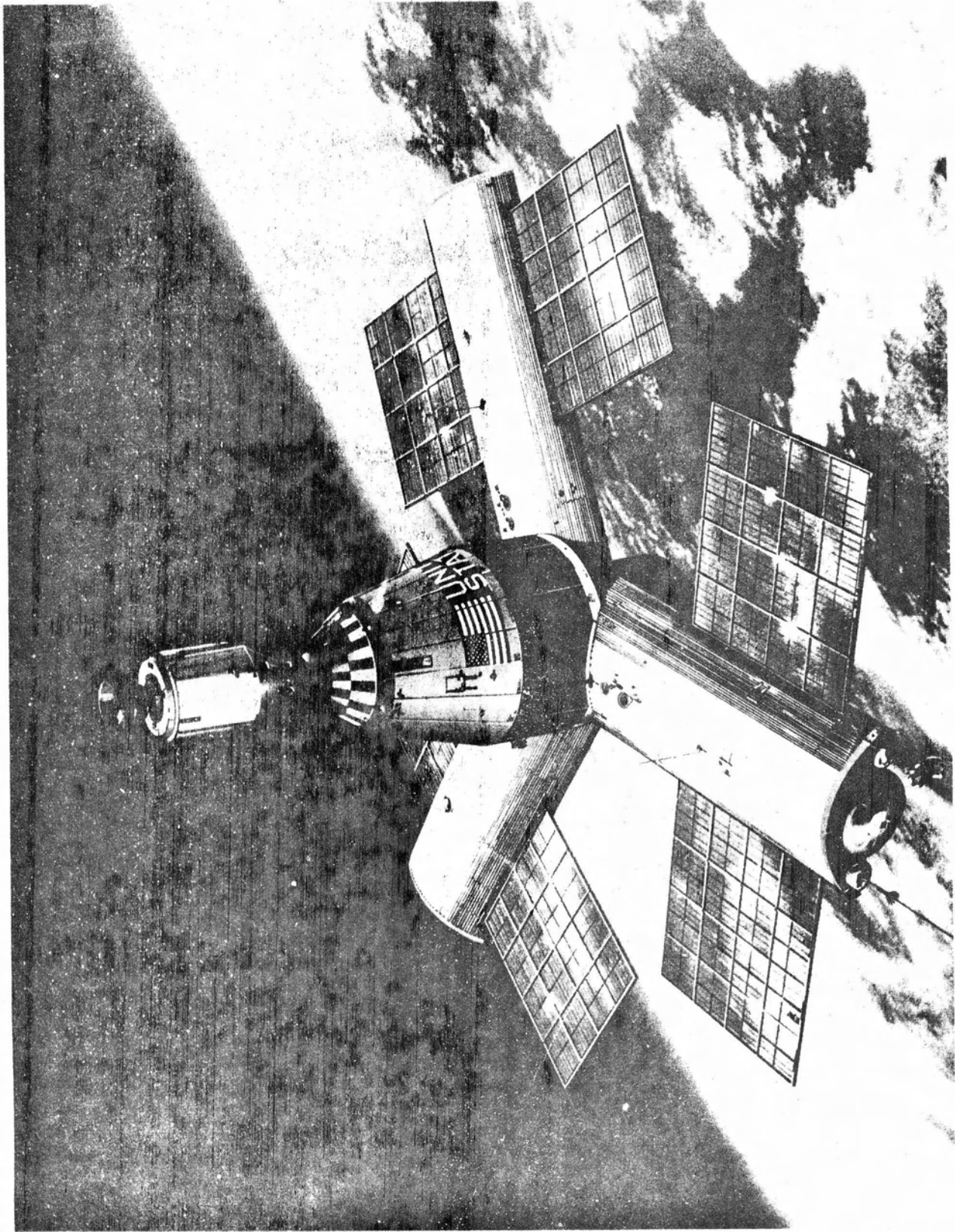
APOLLO LOGISTICS VEHICLE FOR SMSS



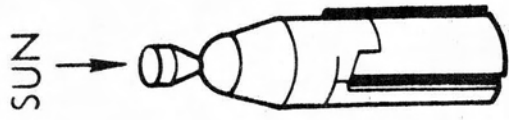
Modified Apollo (MODAP)
Cargo/Crew Modules



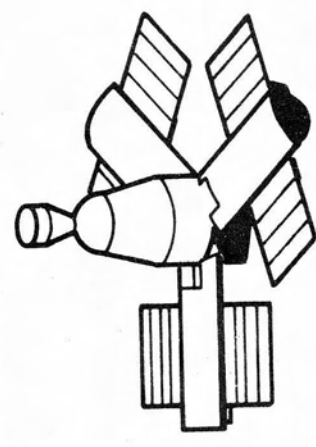
SMSS/Apollo LS Program Schedule



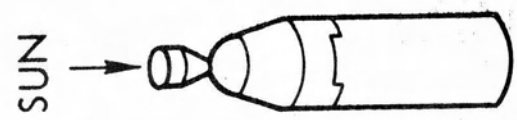
Multi-Module Space Station



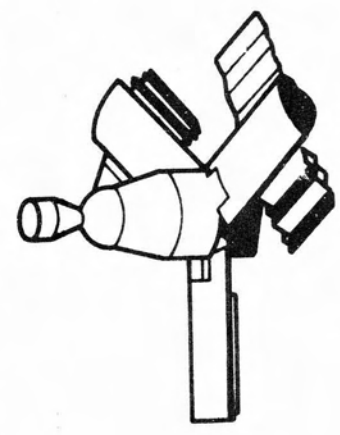
SEPARATE SHROUD SEGMENTS



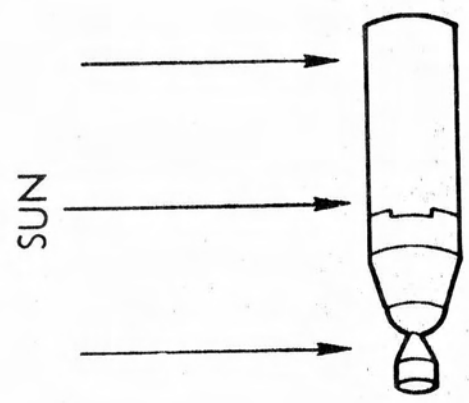
SPIN-UP



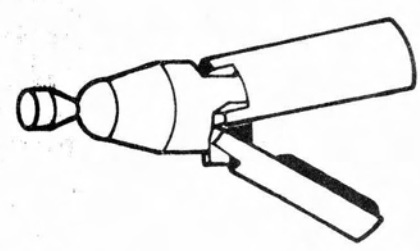
RE-ORIENT



DEPLOY SOLAR ARRAYS

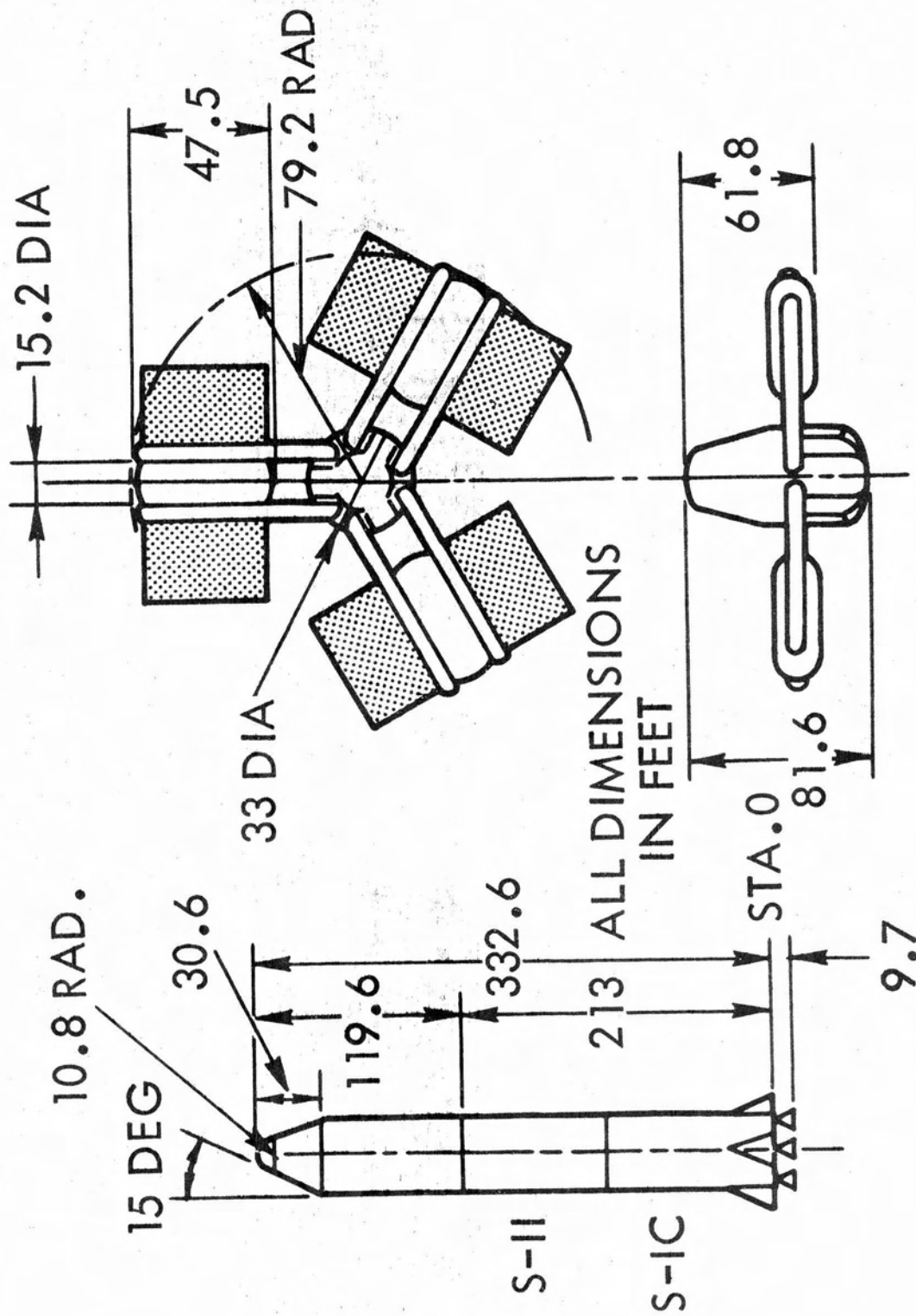


DOCK LS



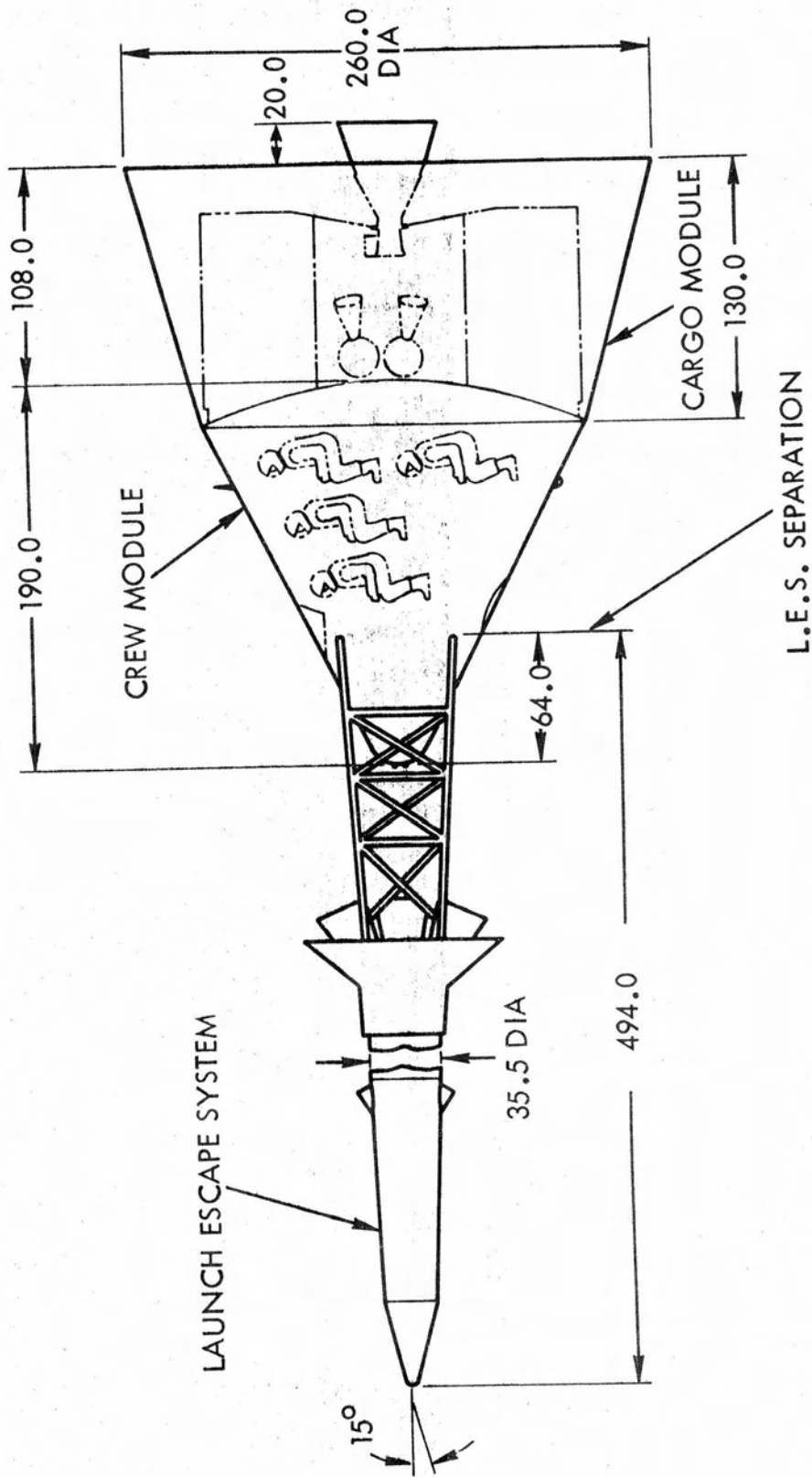
DEPLOY MODULES

Deployment Sequence

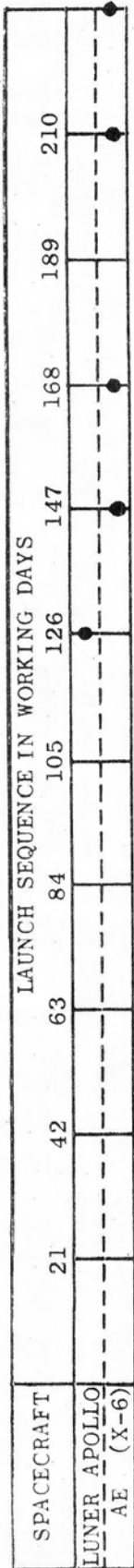


General Arrangement of MMSS

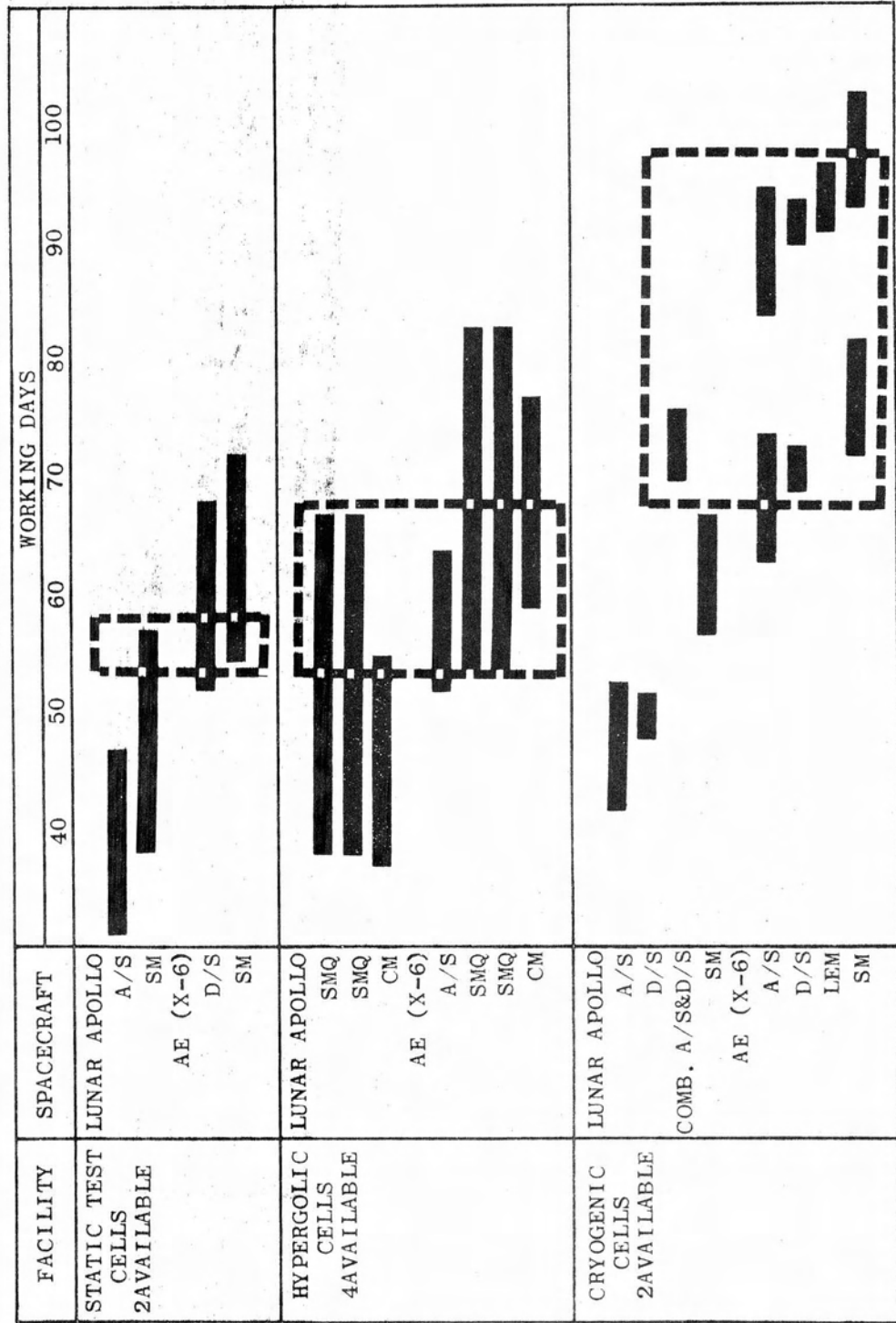
GENERAL ARRANGEMENT



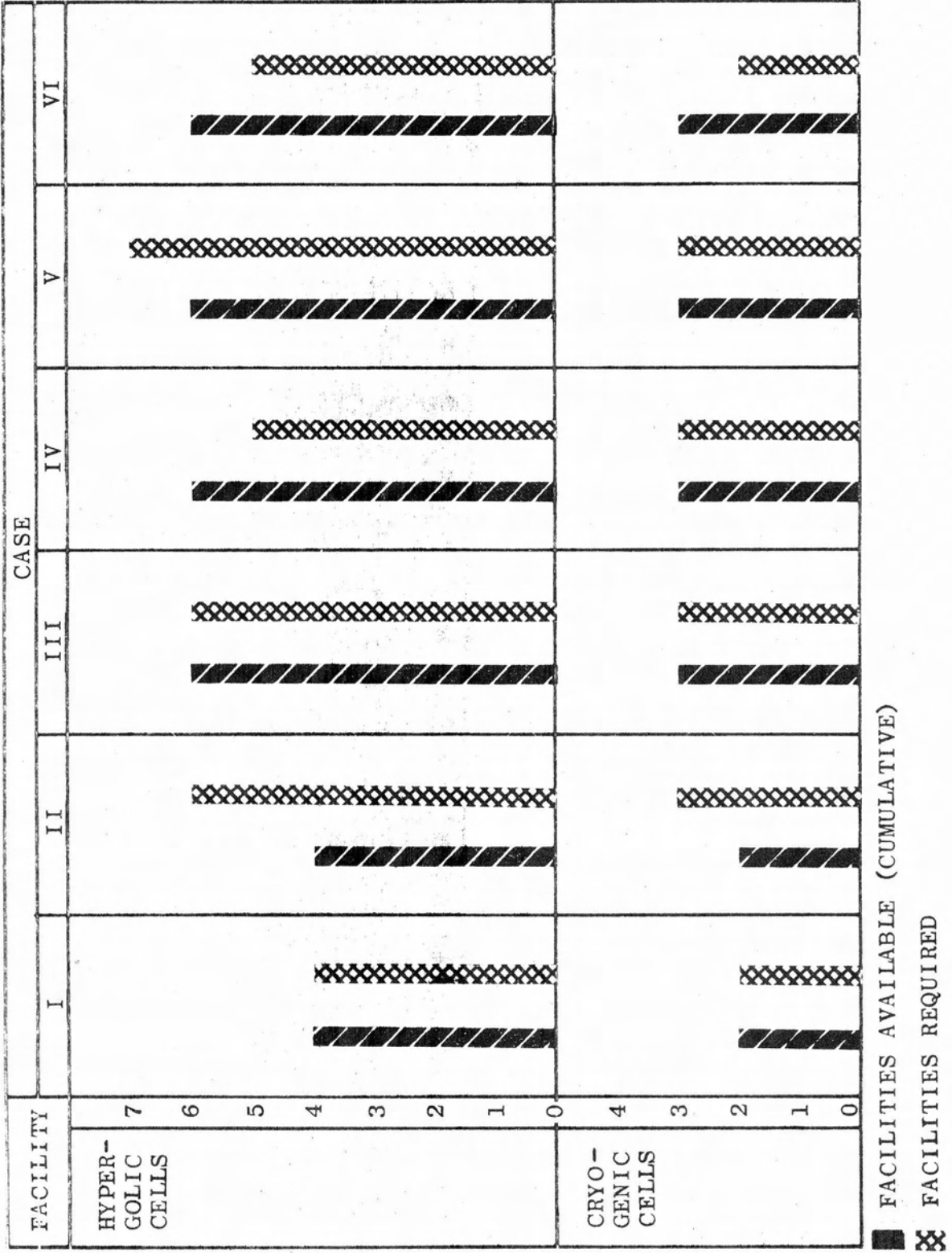
Twelve-Man Ballistic LS — General Arrangement



FACILITY OVERLOAD PERIODS



FLUID TEST COMPLEX FACILITIES REQUIREMENTS



CM PREPARATION

DEGRADATION INSPECTION

COMPLETE DISASSEMBLY

REFURBISHMENT

Electrical
Mechanical
Structural

REASSEMBLY

INSPECTION

S/C CHECKOUT FLOW

CREW MODULE REFURBISHMENT OPERATIONS

BALLOS Refurbishment

Estimated Net Savings

Million Dollars

\$166.5

15 new CM's @ \$11.1 million each

Reuse Concept Costs:

73.6

CM refurbishment

2.0

Refurbishment facility resources

30.7

Reuse RD&T

34.5

Reuse flight test verification

140.8

Potential savings by refurbishment

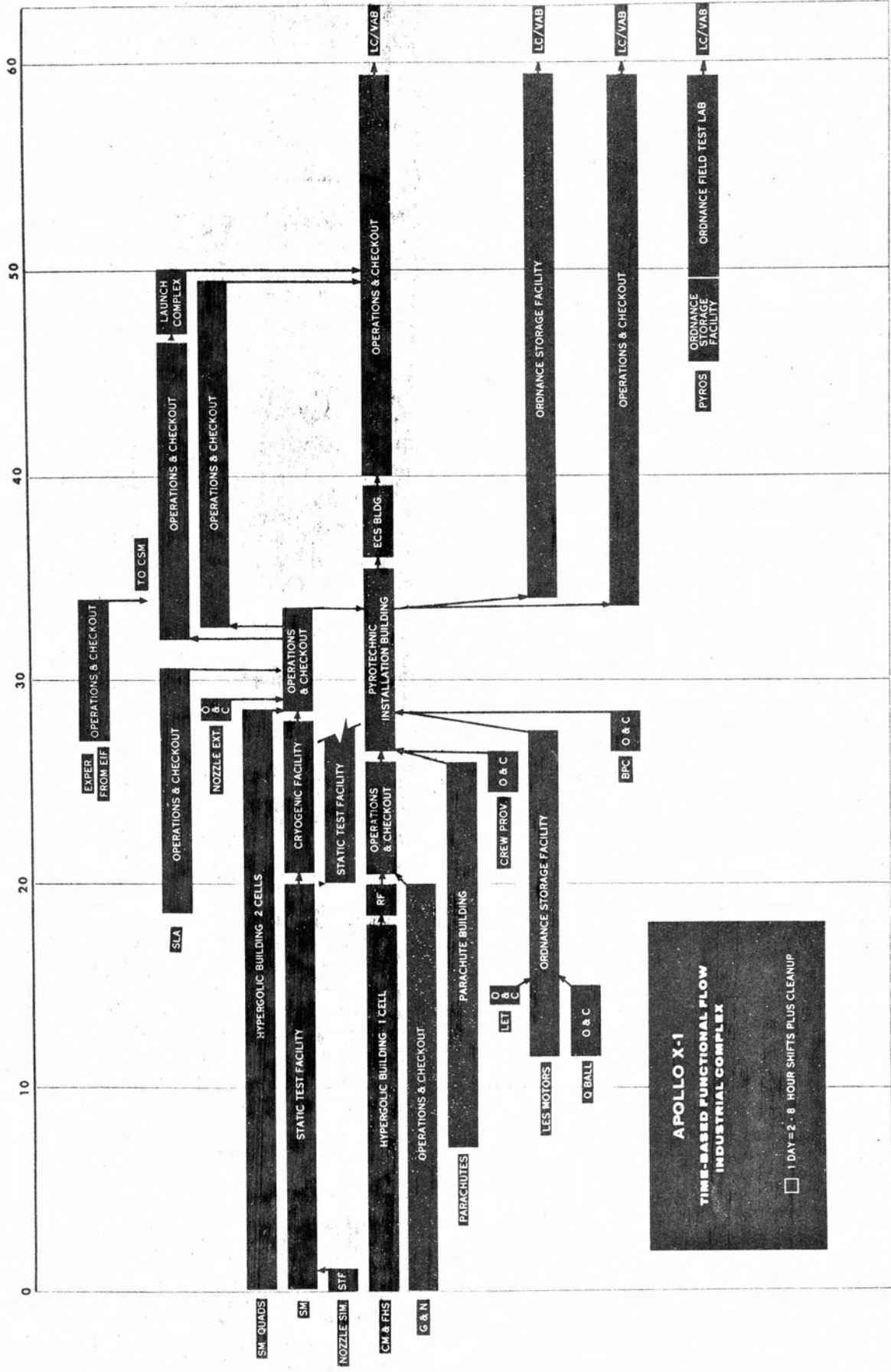
\$ 25.7

Net

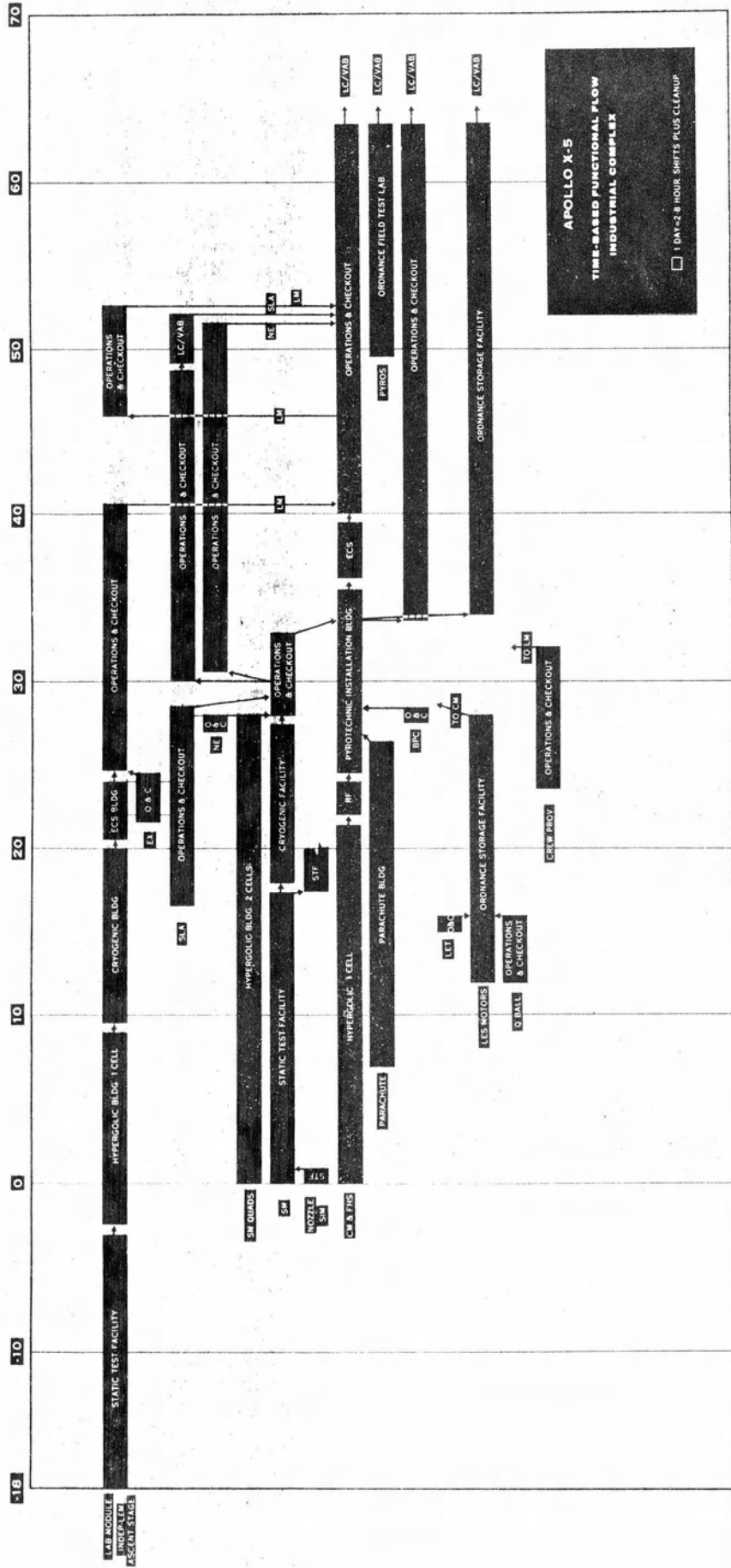
RECOMMENDED MODES OF OPERATION

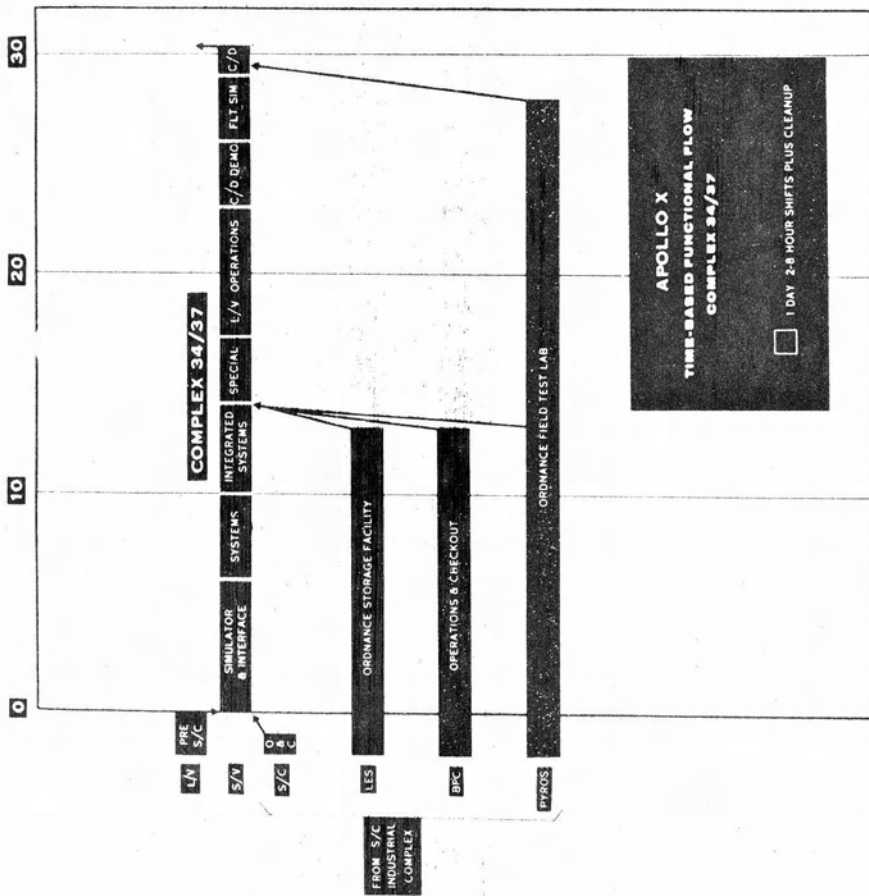
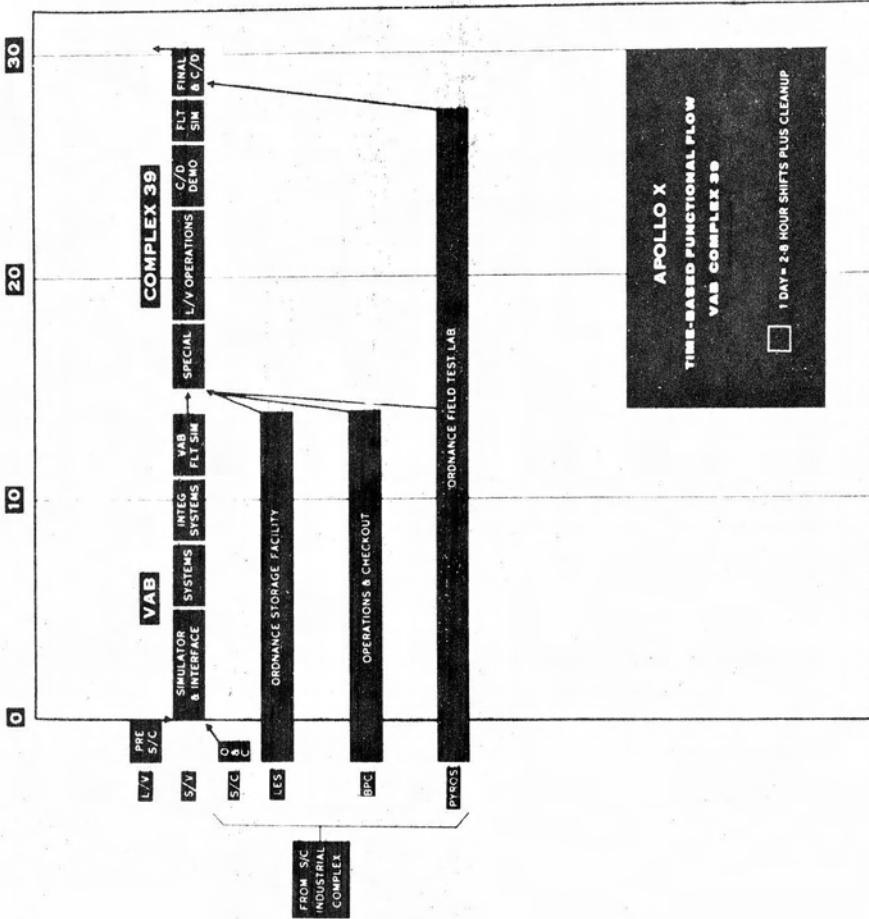
As a product of the subtask and task analyses and of the optimization and tradeoff studies, recommended modes of operation were developed for each of the study configurations.

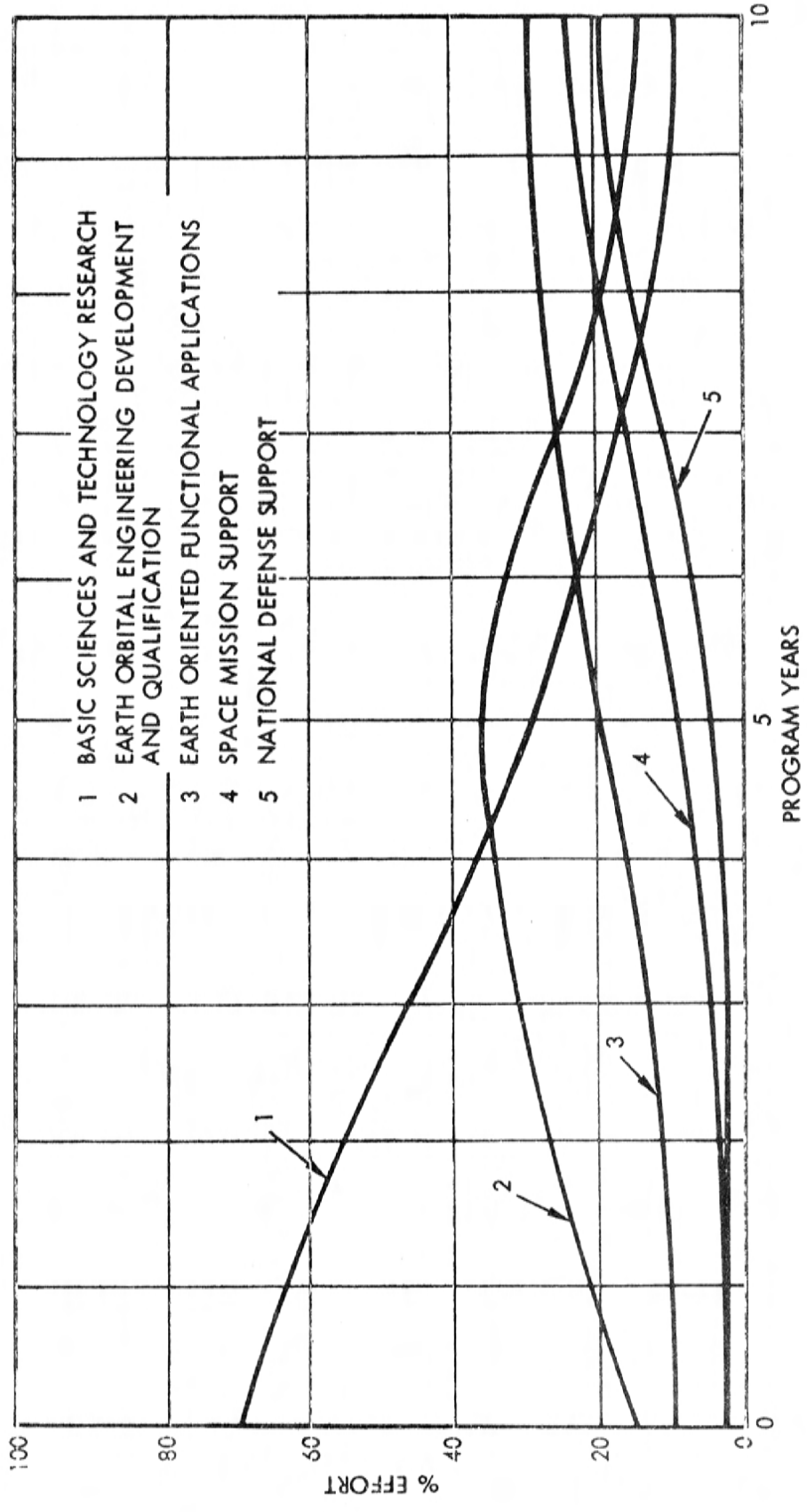
The time-based functional flow diagrams representing the recommended mode of operation for the assembly, checkout and launch of the various study configurations are a composite of functional flow and time line diagrams. The time scale on these diagrams is read horizontally and is in working days. A two-shift operation is assumed with third shift cleanup. No contingencies are considered. Major spacecraft elements are identified vertically and the operations relating to a given module may be identified by locating the module on the vertical listing and following it through horizontally. Similarly, facility utilization at any given time may be established by reading facility and spacecraft elements vertically.











Distribution of Utilization Effort Within ORL Program

EXPERIMENT
SOURCES

UNIVERSITIES

CONTRACTORS

GOVERNMENT
AGENCIES

FOREIGN
SOURCES

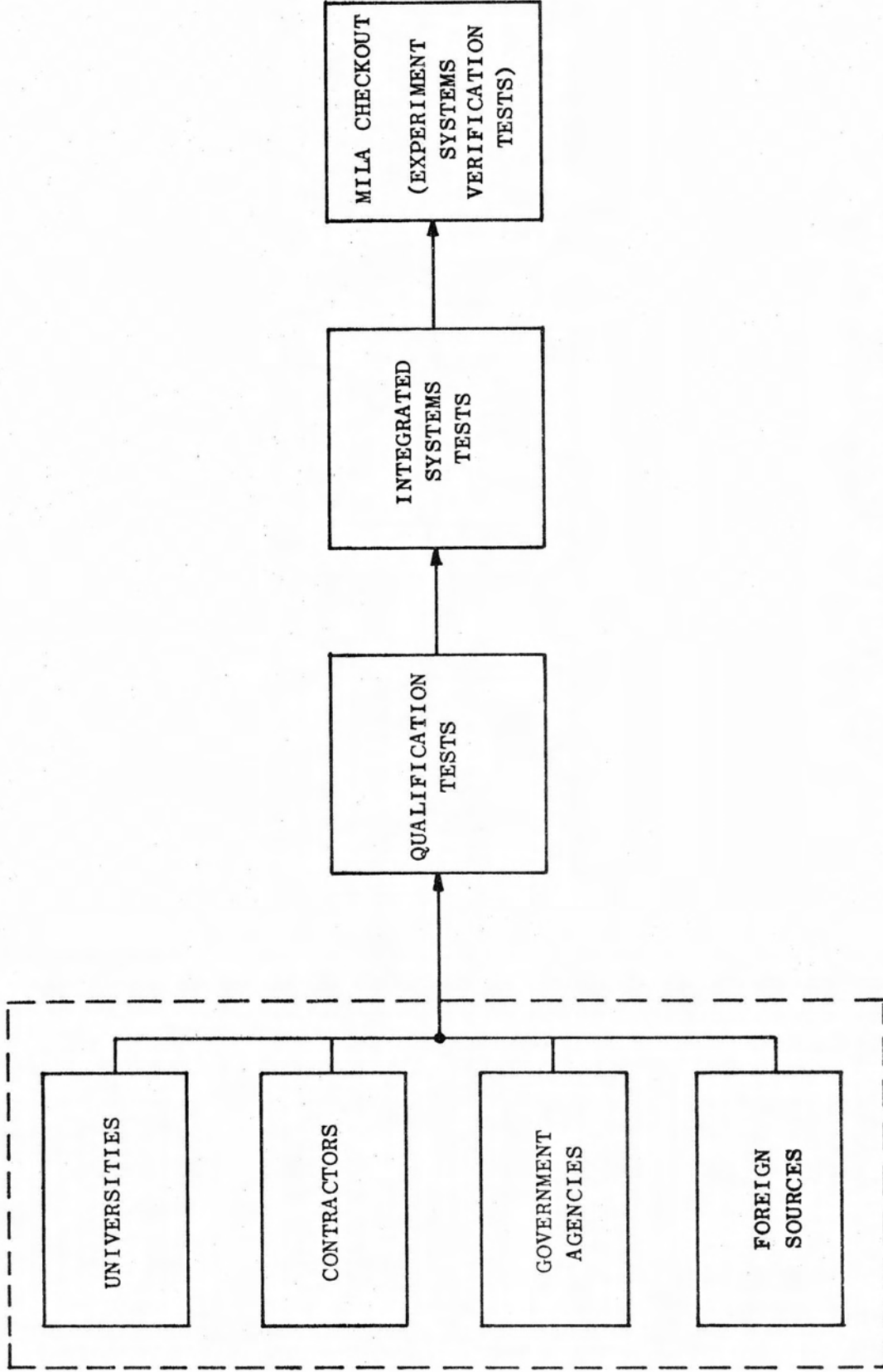
FUNCTIONAL
TESTS

QUALIFICATION
TESTS

INTEGRATED
SYSTEMS
TESTS

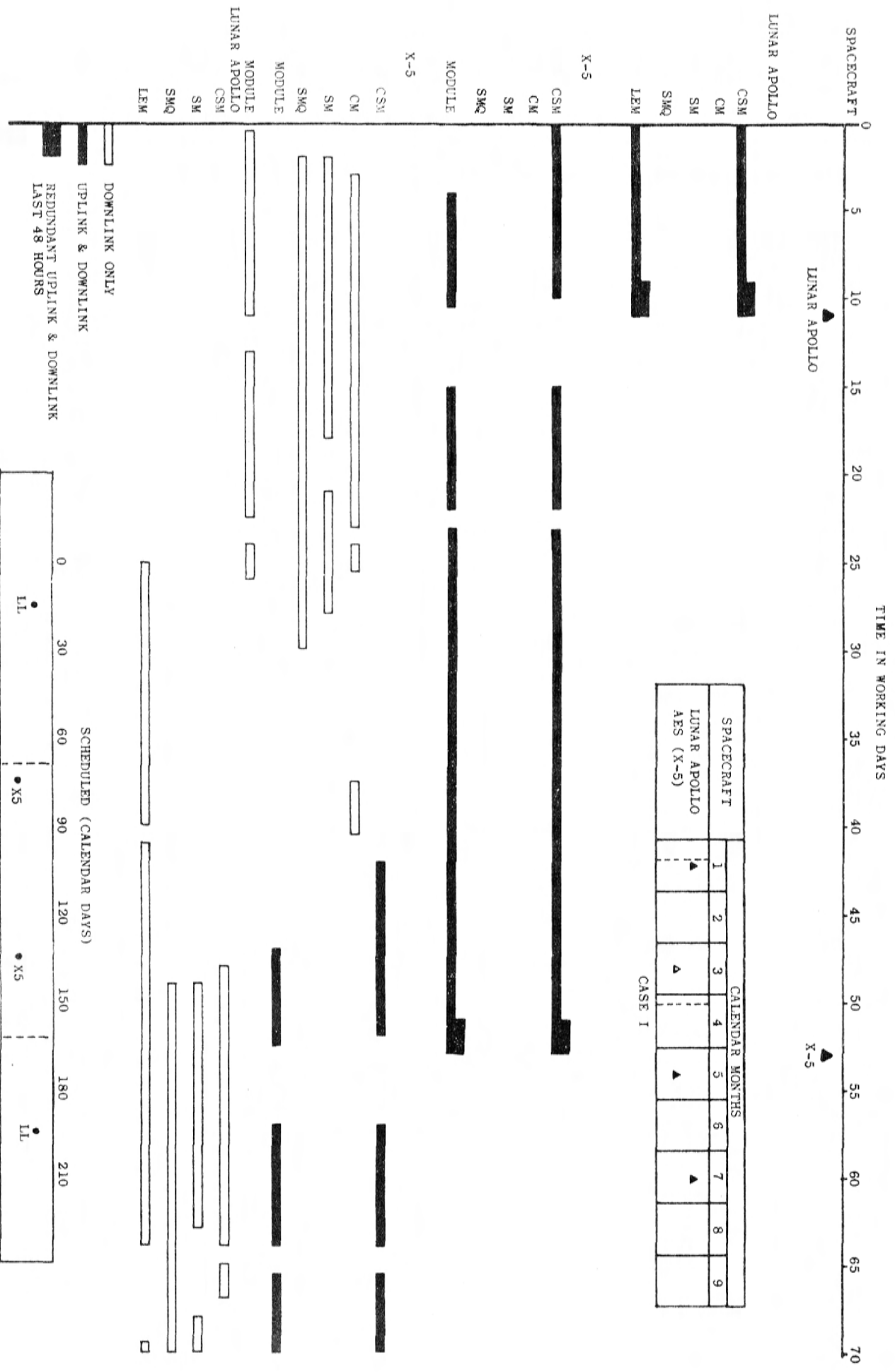
MILA CHECKOUT
(EXPERIMENT
SYSTEMS
VERIFICATION
TESTS)

EXPERIMENT TEST SEQUENCE



ACE ASSUMPTIONS

- Non-interference of Lunar Apollo operations by ORL operations.
- No contingency time is considered.
- Four (4) ACE stations are presently planned, two (2) for spacecraft support and two (2) for LEM support.
- The four (4) ACE stations could easily be modified to support spacecraft subsequent to Lunar Apollo requirements.
- ACE monitoring (downlink only) is required for system level tests.
- Contractor supplied GSE will be used to control the system level tests.
- Complete ACE stations (uplink and downlink) will be used for all tests starting with altitude chamber tests.
- Redundant ACE stations are used during the last 48 hours of the launch area operations.
- An ACE station will be committed to support a specific spacecraft from a specified time subsequent to altitude chamber tests.
- There is no ACE requirement subsequent to launch.
- Post recovery tests and mockup support have not been taken into consideration.
- The instrumentation of all configurations will be compatible with ACE equipment.

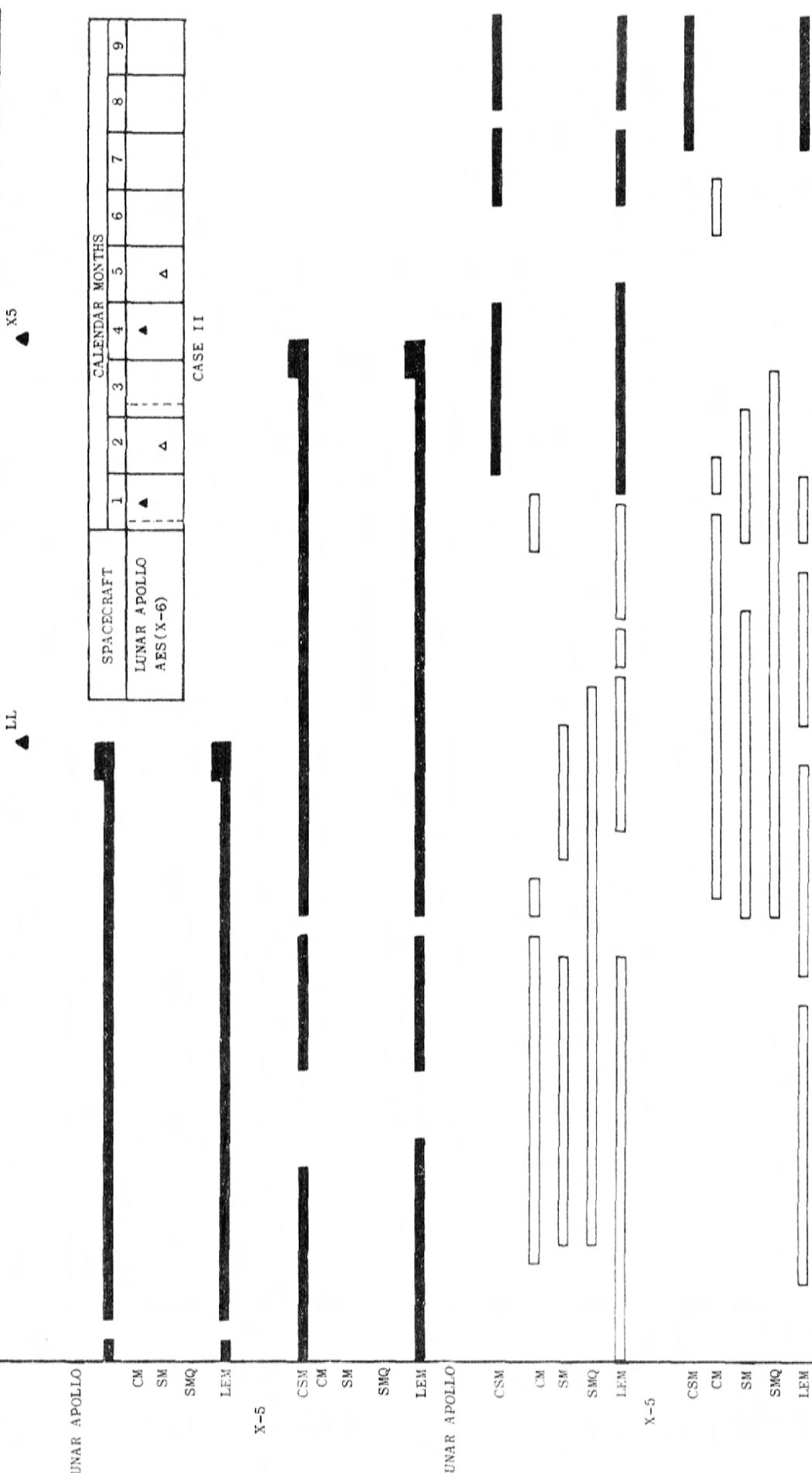


SPACECRAFT	CALENDAR MONTHS								
	1	2	3	4	5	6	7	8	9
LUNAR APOLLO AES (X-5)	▲		▲		▲		▲		

CASE I

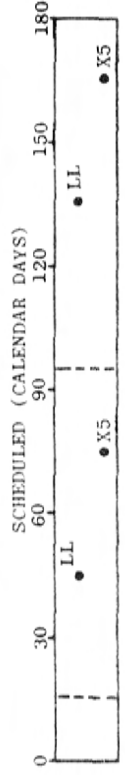
SCHEDULED WORK DAY

0 5 10 15 20 25 30 35 40 45 50 55 60 65 70



SPACECRAFT	CALENDAR MONTHS									
LUNAR APOLLO	▲			▲						
AES (X-6)		▲			▲					

CASE II



DOWNLINK ONLY
 UPLINK ONLY
 REDUNDANT UPLINK & DOWNLINK LAST 48 HOURS

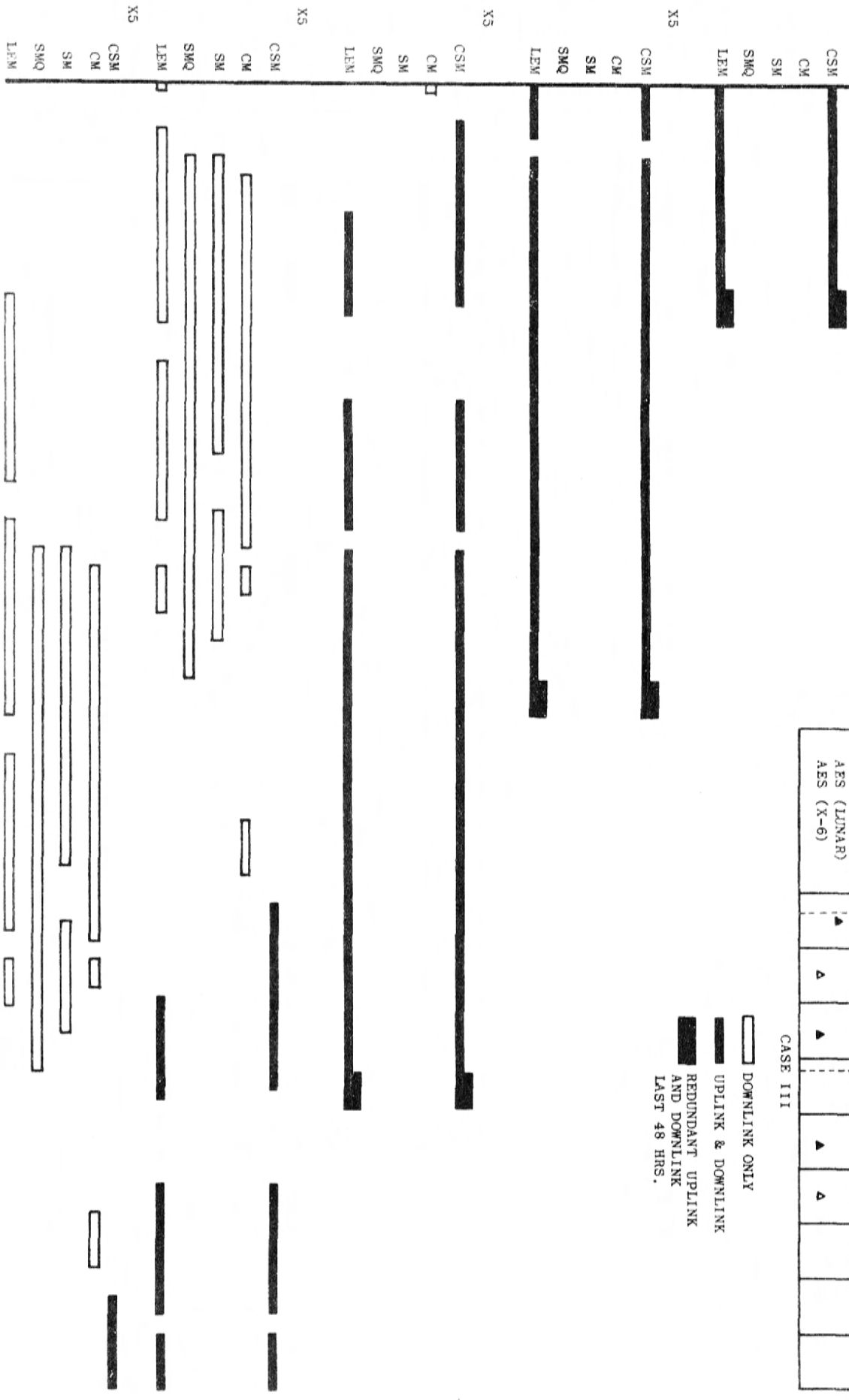
SPACECRAFT 0 5 10 15 20 25 30 35 40 45 50 55 60 65 70

▲ LL

▲ X-6

▲ X-6

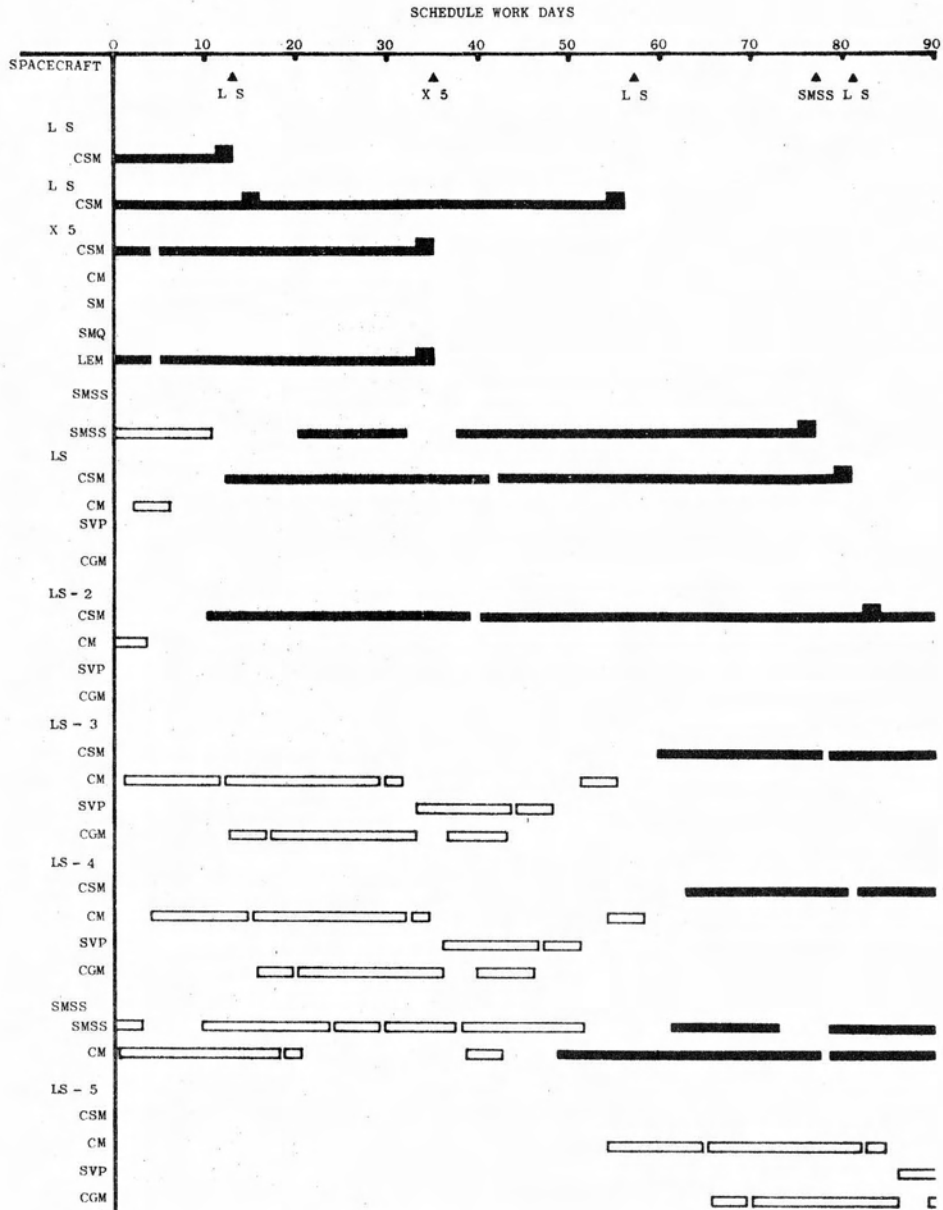
LUNAR APOLLO



SPACECRAFT	CALENDAR MONTHS								
	1	2	3	4	5	6	7	8	9
AES (LUNAR)	▲	▲	▲		▲	▲			
AES (X-6)									

CASE III

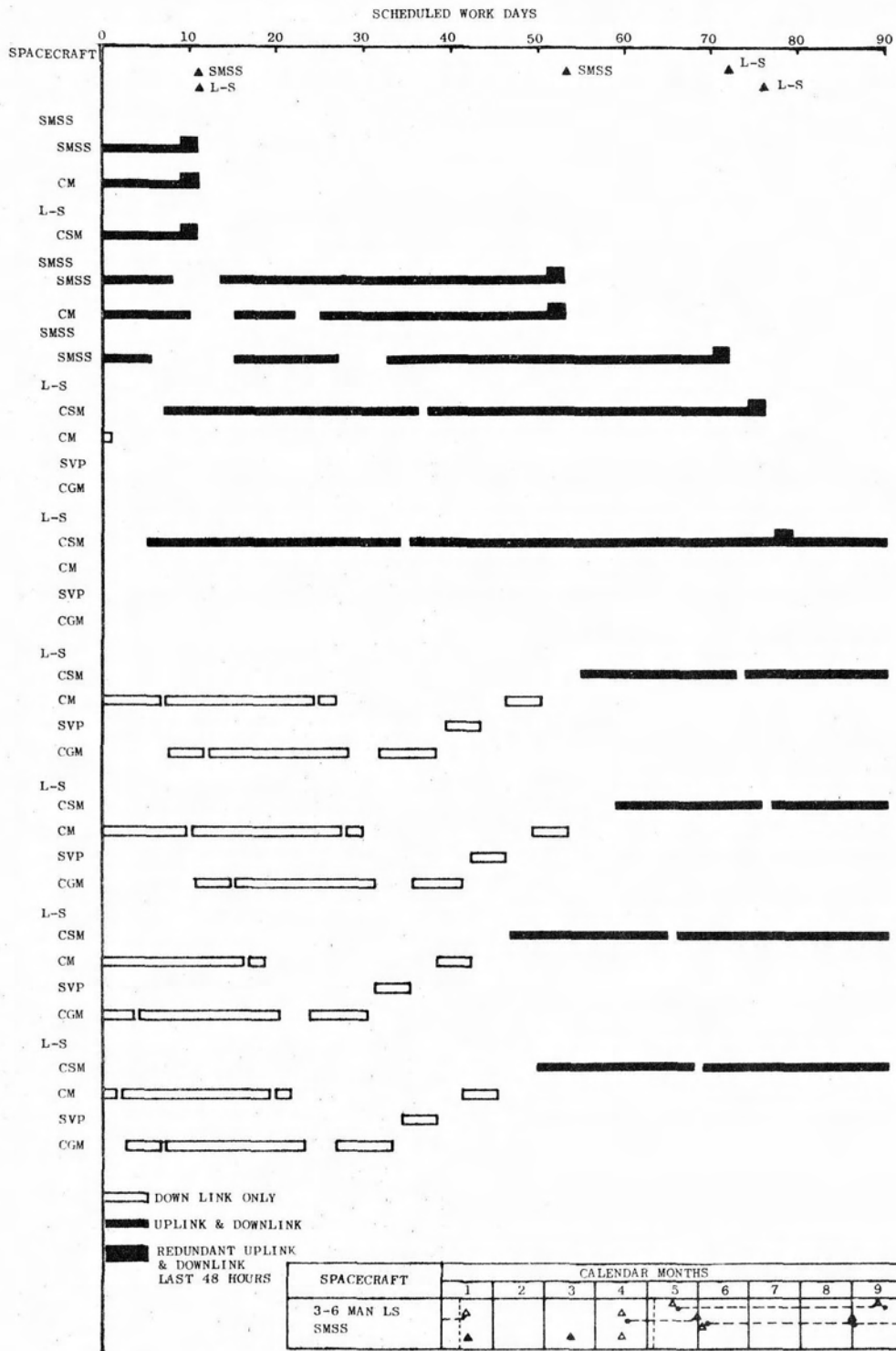
□ DOWNLINK ONLY
 ■ UPLINK & DOWNLINK
 ■ REDUNDANT UPLINK
 AND DOWNLINK
 LAST 48 HRS.



□ DOWNLINK ONLY
 ■ UPLINK & DOWNLINK
 ■ REDUNDANT UPLINK & DOWNLINK LAST 48 hrs.

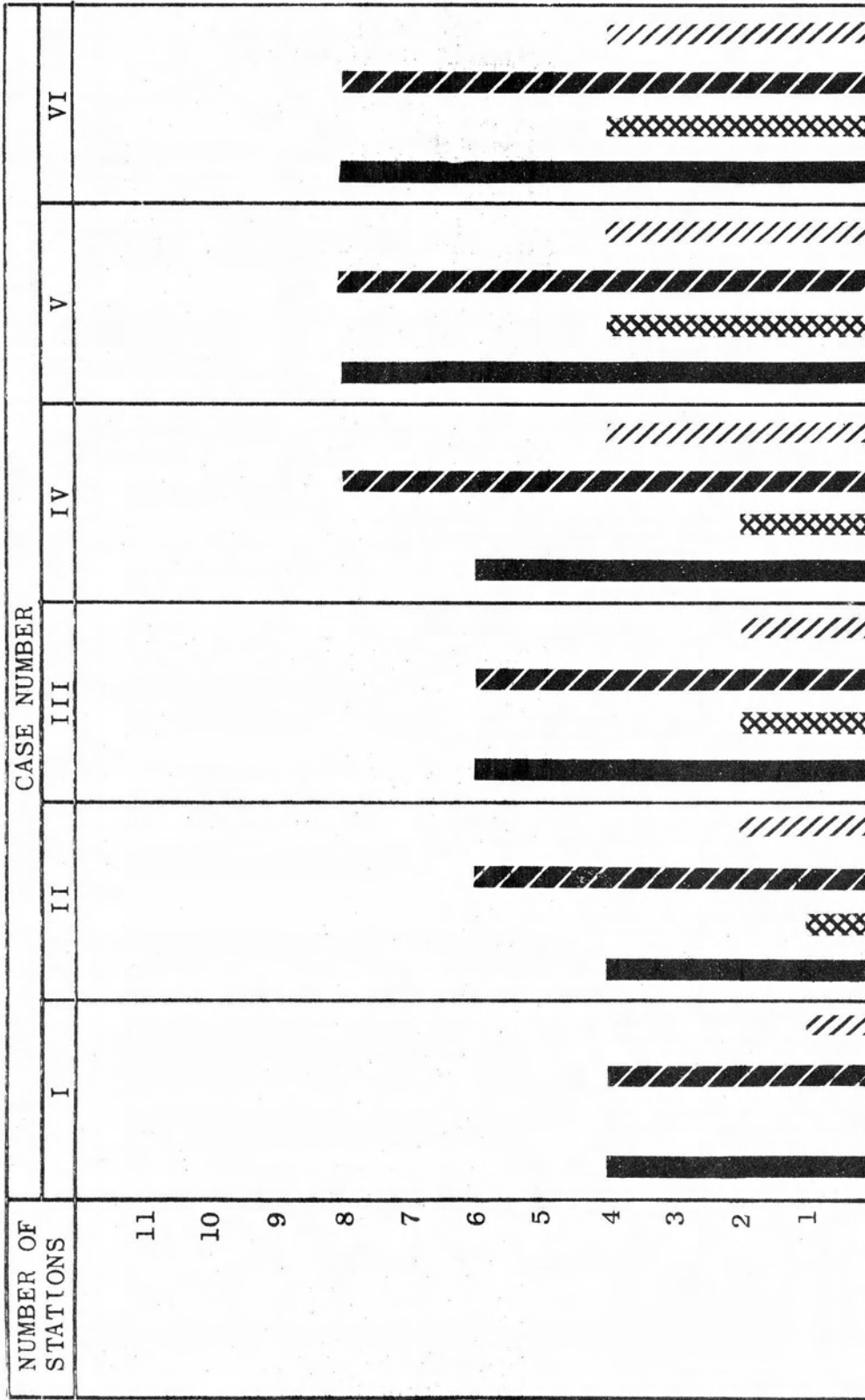
SPACECRAFT	CALENDAR MONTHS								
	1	2	3	4	5	6	7	8	9
AES (X-6)	▲			▲					
3-6 MAN LS			▲	▲	▲	▲		▲	
SMSS						▲		▲	

CASE IV

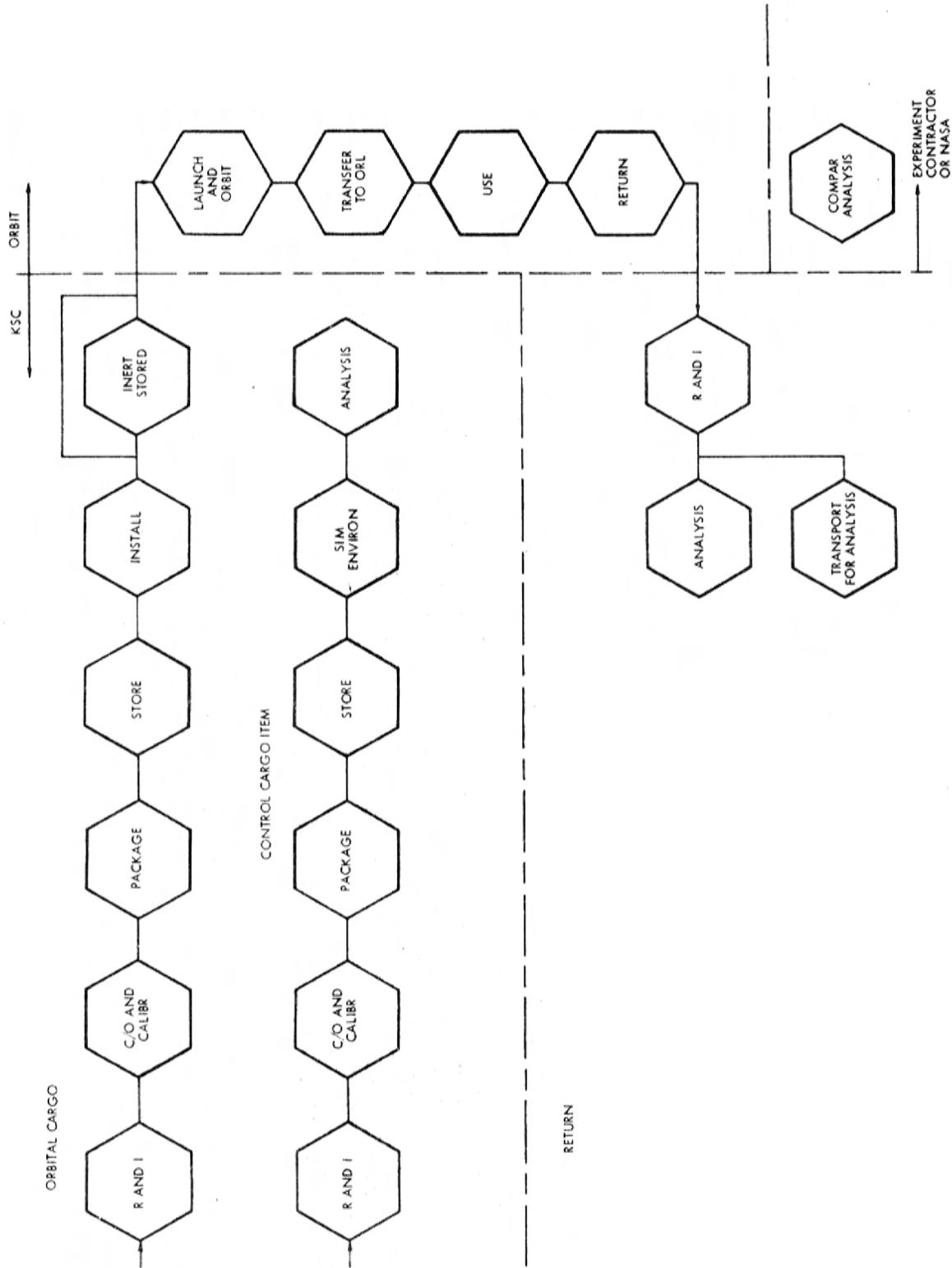


CASE V

ACE REQUIREMENTS



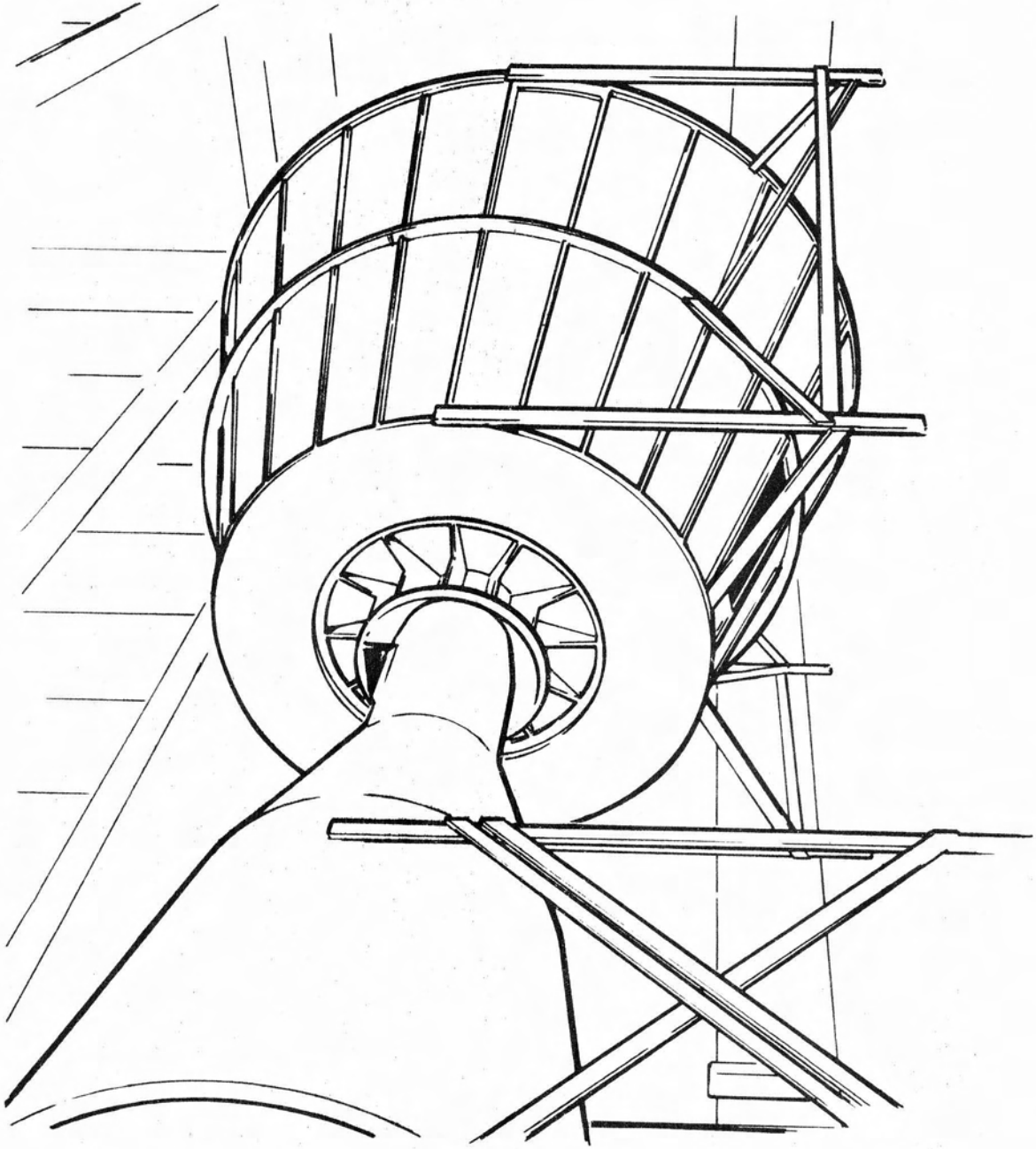
- COMPLETE ACE STATIONS AVAILABLE (CUMULATIVE)
- ▣ DOWNLINK ACE STATIONS AVAILABLE (CUMULATIVE)
- ▤ COMPLETE ACE STATIONS REQUIRED
- ▥ DOWNLINK ACE STATIONS REQUIRED



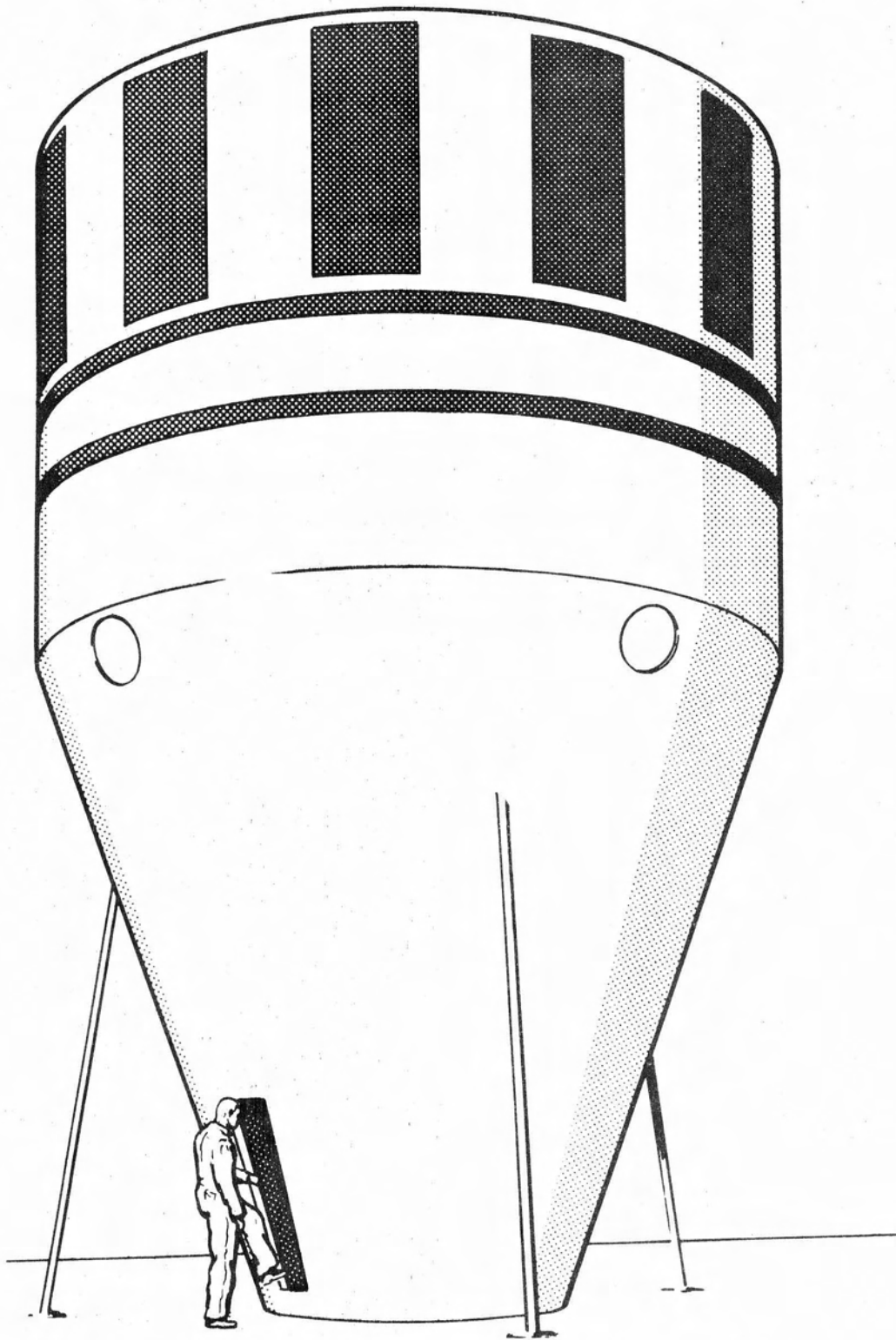
ORL Control Cargo

USE OF
ORL/LS ^{AS} MOCKUPS
TRAINING DEVICES

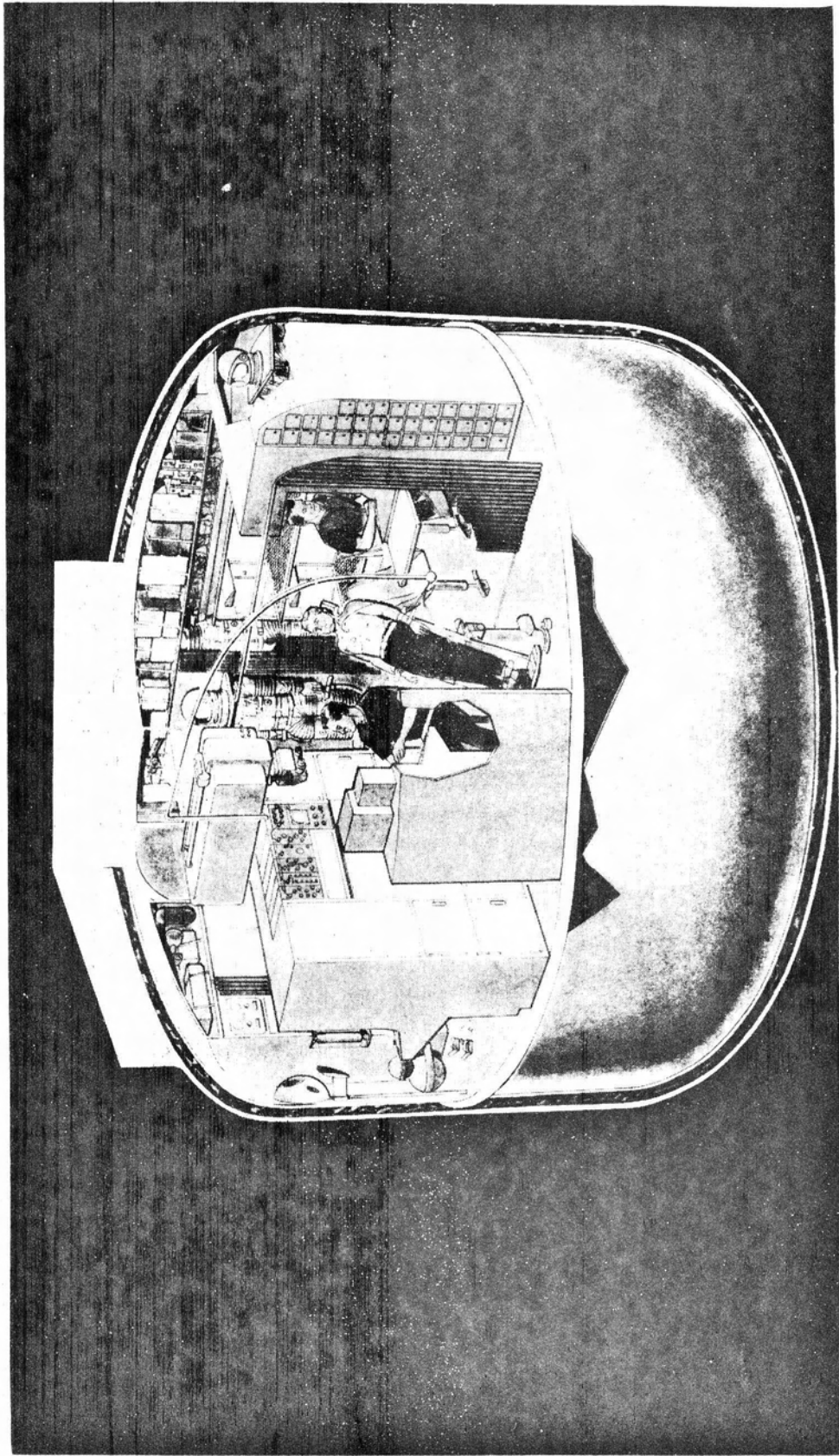
1. LABORATORY MODULE MOCKUP
2. CREW/CARGO TRANSFER MOCKUP
3. CREW MODULE MOCKUP
4. SPARE ORL/LS
5. APOLLO EXTENSIONS
6. SINGLE MODULE SPACE STATION (MORL)
7. APOLLO LOGISTICS SPACECRAFT
8. MODAP LOGISTICS SPACECRAFT



Crew/Cargo Transfer Mockup



MORL Full Scale Mockup

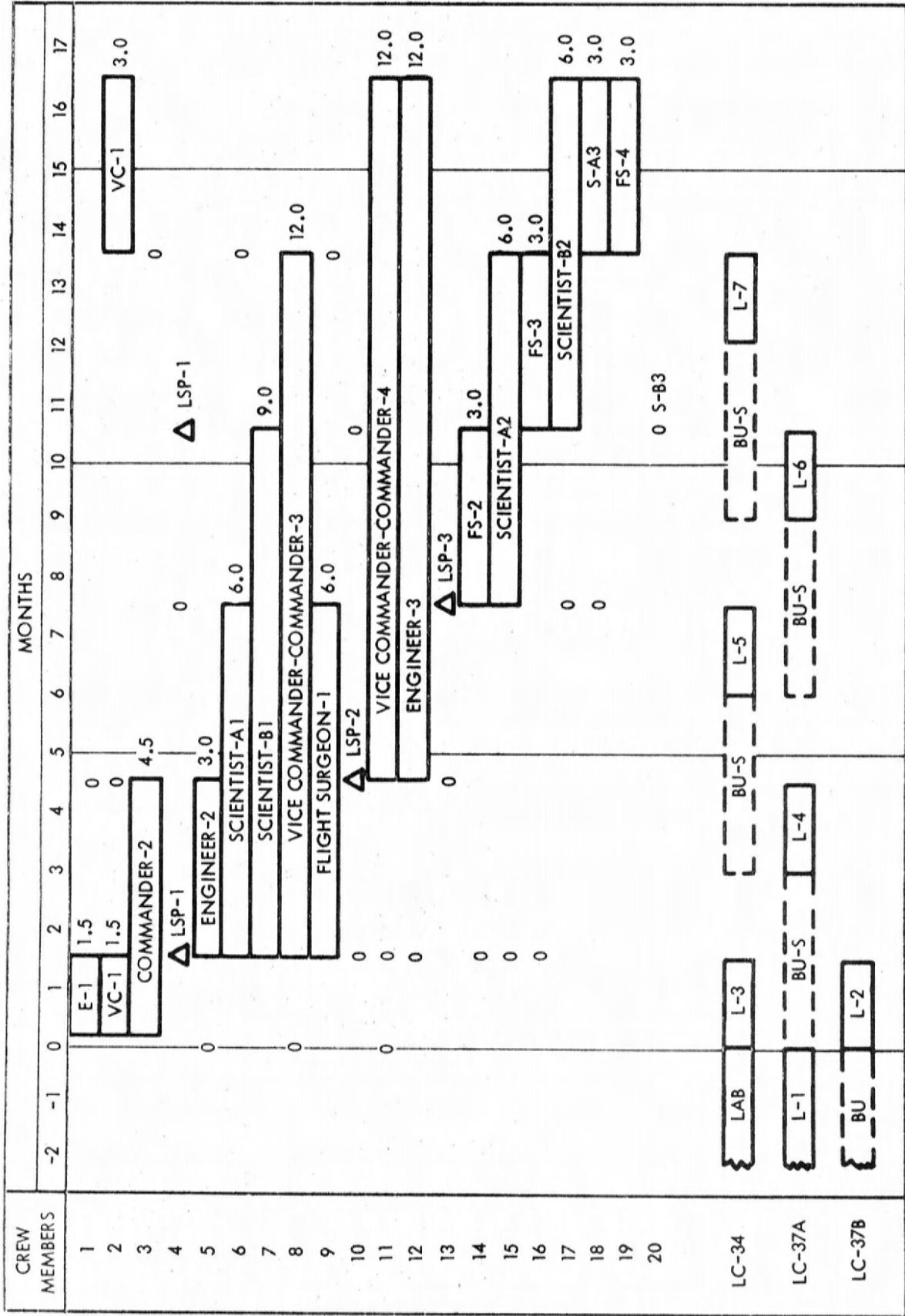


MMSS Zero-G Laboratory Upper Level Mockup

CREW MEMBERS	MONTHS																																					
	0	1	2	3	4	5	6	7	8	9	10	11	12	13	14	15	16	17	18	19	20	21	22	23	24	25	26	27	28	29	30	31	32	33	34			
1																																						
2																																						
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11																																						
12																																						
LAUNCH COMPLEX																																						
LC-37																																						
LC-39																																						

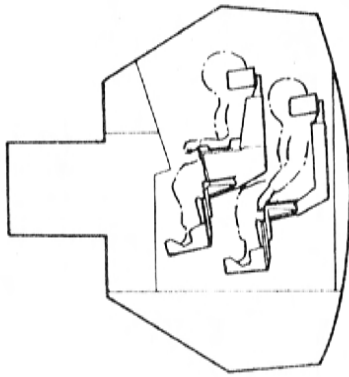
0 - BACKUP CREW
 F - FLIGHT OPERATIONS SKILL
 S - FLIGHT SURGEON

Typical Apollo Extensions Mission and Backup Crew Provisions
 Apollo X-2 through X-6

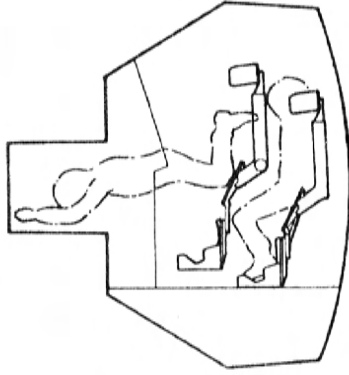


S - A3 - SCIENTIST SKILL A NUMBER 3
 0 - BACKUP
 L - LOGISTICS SPACECRAFT
 Δ LSP - LOGISTICS SPACECRAFT PILOT
 BU-S - BACKUP - STANDBY

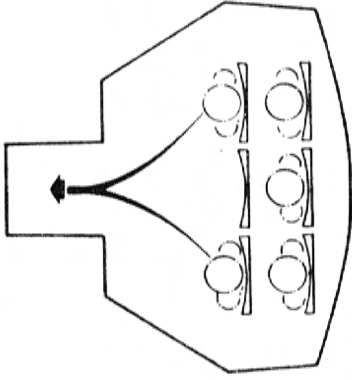
Typical SMSS/Apollo-LS Mission and Backup Crew Provisions



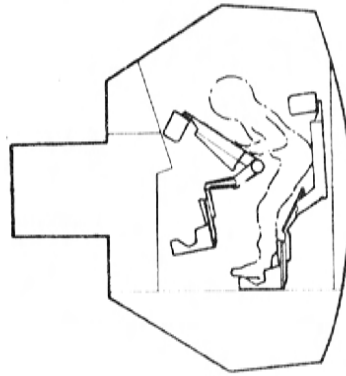
1. ALL OCCUPANTS LOWER SEATS



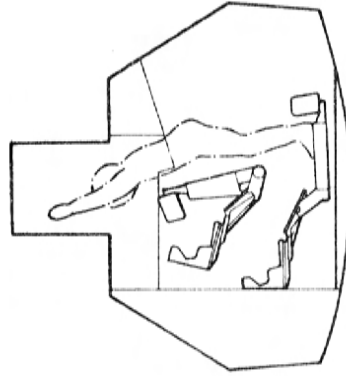
2. UPPER CENTER REMOVES & STOWS DOCKING HATCH - MAKES EXIT



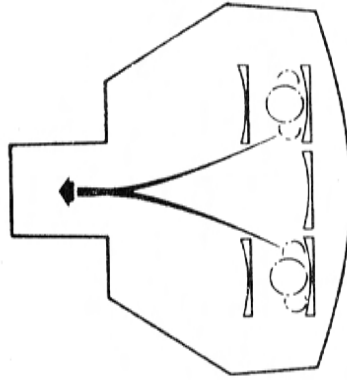
3. UPPER OUTSIDE ROLL TO CENTER & EXIT



4. LOWER CENTER FOLD UPPER CENTER SEAT

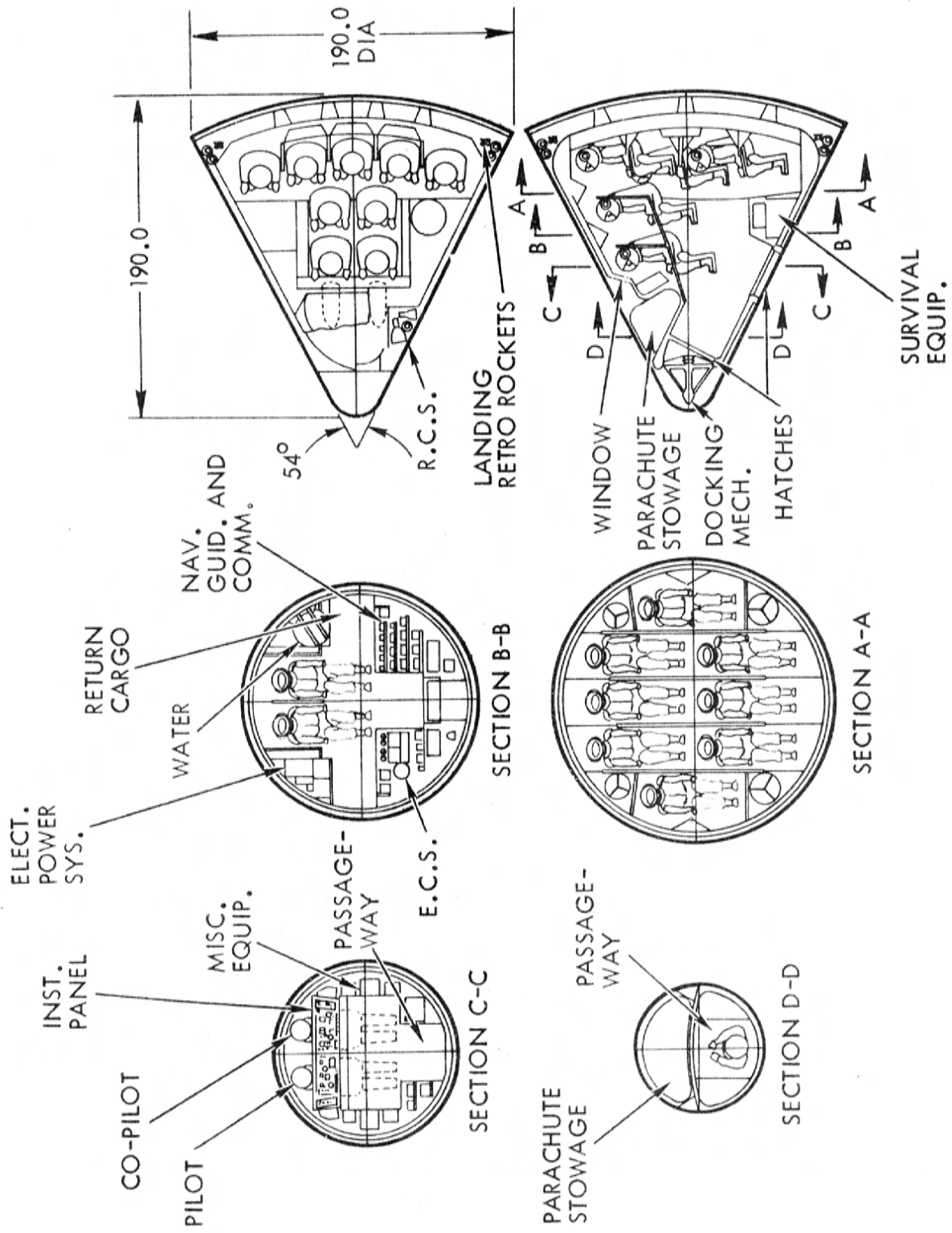


5. LOWER CENTER EXITS

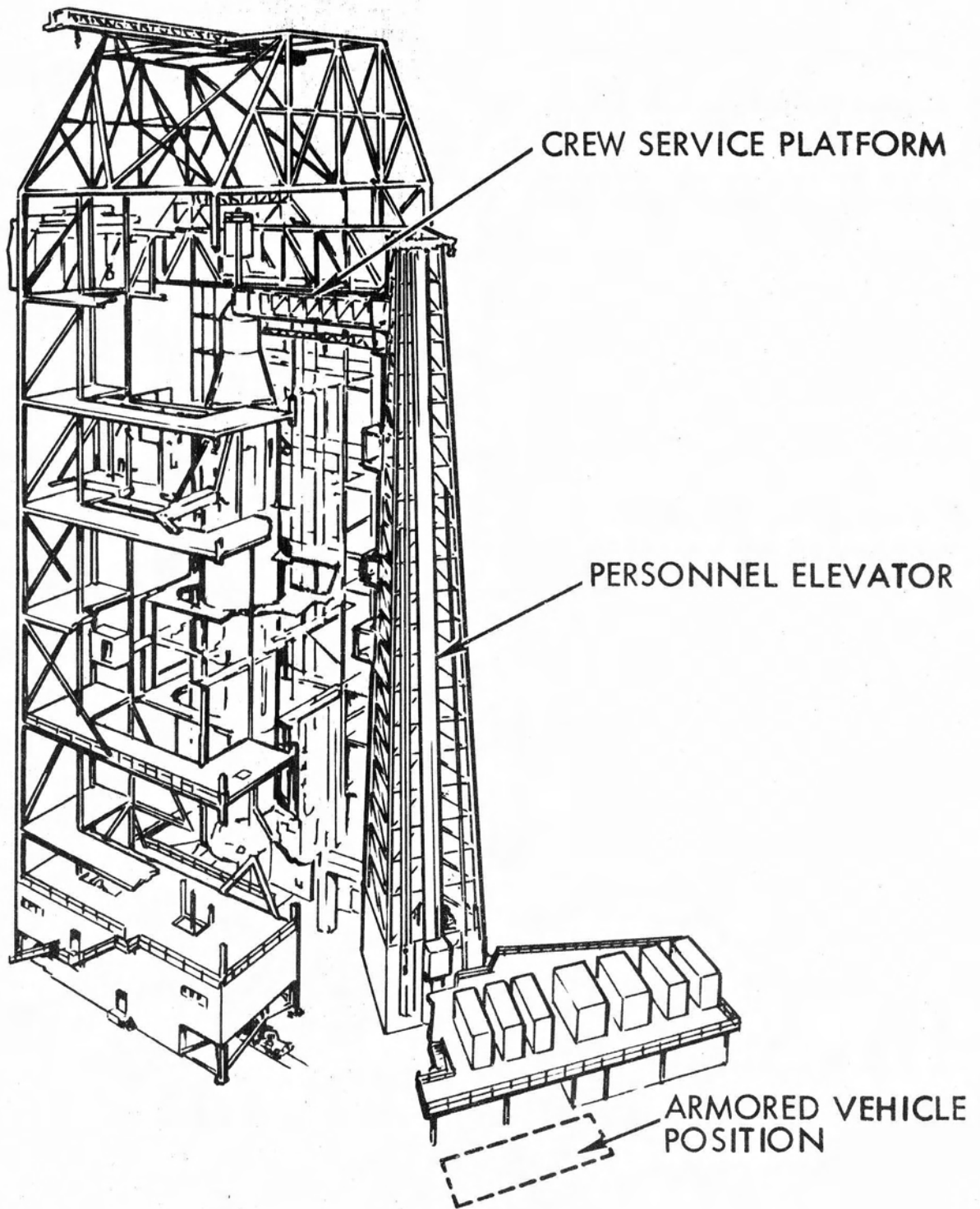


6. LOWER OUTSIDE SLIDE TO CENTER & EXIT

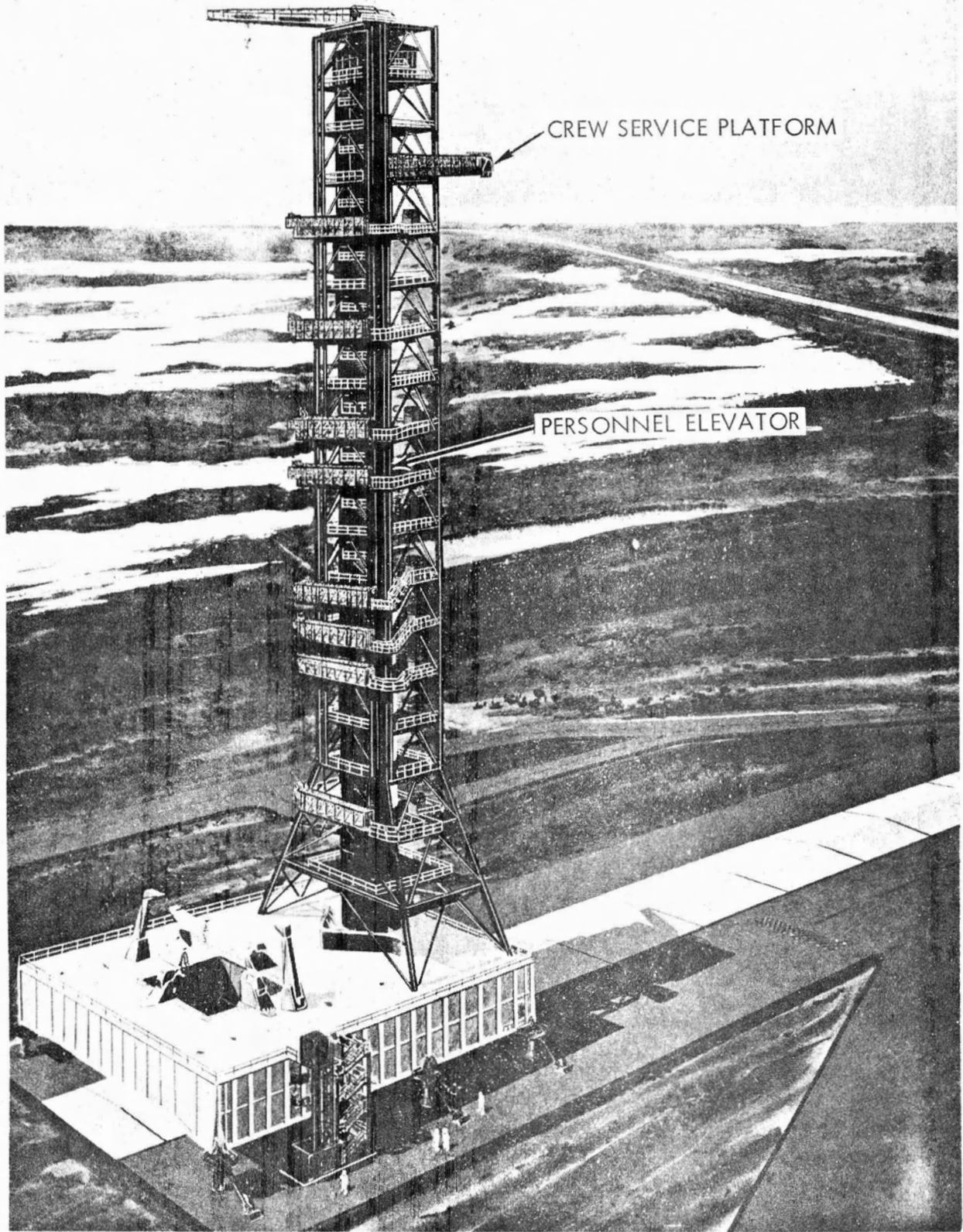
MODAP Exit Procedure



Crew Module Details



Launch Complex 34 Emergency Removal Provisions



Launch Complex 39 Emergency Removal Provisions

BUDGET IMPACT SUMMARY
(MILLIONS OF DOLLARS)

ITEM	CASE					
	I	II	III	IV	V	VI
FACILITIES	0.4	3.4	0.1	7.6	0.1	6.1
EQUIPMENT	2.8	14.2	—	17.4	—	1.5
TOTAL	3.2	17.6	0.1	25.0	0.1	7.6

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CONCLUSIONS

- Cryogenic and hypergolic test cells are adequate in size and function to support AE study configurations.
- Additional cryogenic and hypergolic test cells required to support AE study program baseline schedule.
- Larger cryogenic and hypergolic test cells required to support SMSS & MMSS study program baseline schedule.
- Modifications to ECS test cell required to support SMSS and MMSS study configurations.
- Additional ACE stations required to support ORL/LS study program baseline schedule.

CONCLUSIONS

- Experiment integration and test capability required to support ORL/LS study experiment program.
- Cargo checkout and packaging facility required to support SMSS study program resupply.
- Integrated checkout facility required to support SMSS study configuration.
- Assembly and checkout facility required to support MMSS study configuration.
- Altitude chambers not large enough for MMSS study configuration.
- Additional crew quarters, training facilities and mission simulators required to support ORL/LS study configurations and study program baseline schedule.

CONCLUSIONS

- Detailed configuration control required for resupply cargo and S/C mockups.
- Hazardous checkout operations result from SMSS and MMSS study configurations.
- Proximity of Ordnance Storage Facility to Pyrotechnic Installation Building requires protection from inadvertent ignition of LES motor study configuration.
- Clearance survey of S/C transport routes required by SMSS and MMSS study configurations.
- Refurbishment of BALLOS CM is feasible; new facility required to support refurbishment.

RECOMMENDATIONS

- Provide two additional larger hypergolic cells and one additional larger cryogenic cell to support the ORL/LS study configurations and program baseline schedule.
- Provide 4 down link ACE stations and 4 additional complete ACE stations to support the ORL/LS study program baseline schedule.
- Modify one existing ECS cell to support SMSS and MMSS study configurations.
- Provide experiment integration and test capability to support ORL/LS study experiment program.
- Provide cargo checkout and packaging facility to support SMSS study program resupply requirements.
- Provide integrated checkout facility to support SMSS study configuration.

RECOMMENDATIONS

- Provide assembly and checkout facility to support MMSS study configuration.
- Provide future study of requirement for altitude chamber testing of MMSS study configuration.
- Provide crew quarters, training facilities, and mission simulators to support ORL/LS study configurations and study program baseline schedule.
- Provide detailed configuration control procedures for resupply cargo and spacecraft mockups.
- Redesign SMSS and MMSS spacecraft and associated logistic spacecraft to preclude introduction of hazardous ordnance into normally non-hazardous areas.
- Provide revetment between ordnance support facility and pyrotechnic installation building.

RECOMMENDATIONS

- Survey SMSS and MMSS transport routes for adequate clearance.
- Provide refurbishment facility for BALLOS command module.
- Use operational modes for ORL/LS checkout, experiment integration and test, and crew and cargo operations as a basis for further planning and program definition.
- Integrate requirements of other spacecraft programs with requirements for ORL/LS and lunar Apollo programs to determine total requirements to be placed on MILA resources.
- Study launch vehicle-spacecraft interface and space vehicle-facility and equipment interface to determine additional requirements on MULA resources.
- Study tradeoffs between program baseline schedules and MULA resources utilization, as an input to overall program planning.