

Johnson
Film Scripts

FILM SCRIPT

for

SATURN QUARTERLY FILM REPORT NO. 12

(Covering April, May, June, 1962)

Unclassified

FILM

de in

UNCLASSIFIED Title

Dissolve

First Title:

The

GEORGE C. MARSHALL
SPACE FLIGHT CENTER
Presents

Second Title:

SATURN QUARTERLY FILM REPORT NO. 12
(Covering April, May, June, 1962)

Sequence no. 1:

ight shots of
SA-2 on pad

NARRATION

OPENING MUSIC

At almost midnight on April 24th at Cape Canaveral's Launch Complex 34, the scheduled 10-hour-long countdown began for the launching of the second Saturn flight test vehicle, SA-2.

Sequence no. 2:

Countdown activity

All automatic propellant loading and sequencing processes were conducted satisfactorily.

Sequence no. 3:

More night shots on SA-2

The countdown proceeded without a single technical hold.

FILM

Sequence no. 4:
More blockhouse activity

Sequence no. 5:
SA-2 on pad just before
launch; countdown clock

Sequence no. 6:
SA-2 ignition and lift off

Sequence no. 7:
SA-2 in flight
(several views)

NARRATION

One 30-minute range hold was called, however, until a ship could clear the range area.

Shortly after 9 a.m. on April 25th-- six months after the spectacularly successful first Saturn flight--the countdown for SA-2 had reached its final seconds.

(SOUND EFFECTS: FINAL 10 SECONDS OF COUNTDOWN. SA-2 IGNITION AND LIFT OFF. THEN, LOWER SOUND EFFECTS, AND VOICE OVER--)

Ignition, thrust build-up, and lift-off were normal. Objectives of the SA-2 launch included flight testing of the booster stage and operational testing of associated launch facilities. Structural integrity of the Block I air frame and aerodynamic characteristics were confirmed, and capabilities of the control system demonstrated. The propulsion system performed normally throughout powered flight.

FILM

Sequence no. 8:

cut to shot of men
at telemetry consoles
in blockhouse during
flight

Sequence no. 9:

SA-2 in flight

Sequence no. 10:

Animation of SA-2
launch and flight path
POP ON figure "65 miles"

Animation of explosion

NARRATION

(OMIT PREVIOUS SOUND EFFECTS DURING
THIS SCENE.)

All electrical networks and instrumentation
functioned properly, with very satisfactory
telemetry signals received.

(RESUME BACKGROUND SOUND EFFECTS OF
VEHICLE IN FLIGHT.)

Maximum velocity of over 3,700 miles per
hour was attained. The sloshing effects
observed during the SA-1 flight was reduced
to an acceptable level. Cut-off occurred
at 110 seconds for inboard engines and 116
seconds for outboard, as predicted. In
virtually every respect, the SA-2 flight was
successful.

SA-2 also carried out a secondary, or
"bonus," scientific experiment known as
"Project High Water." At an altitude of
65 miles, the vehicle -- whose dummy
upper stages carried 23,000 gallons of
water as ballast -- was purposely exploded,
to investigate the optical, ionospheric,
and meteorological effects which water
vapor has on the high atmosphere.

FILM

Sequence no. 11:

Film of actual
water burst

Sequence no. 12:

LS of lab set-up,
showing vacuum chamber
and control instrumentation
(P-496); CU of Vacuum chamber,
showing colored water in
small test tube

Sequence no. 13:

Slow motion -- CU of
vacuum chamber, showing
hammer striking
horizontal tube and
water escaping

NARRATION

About 15 per cent of the water evaporated,
and the remaining 85 tons formed a cloud
of very small ice particles along the
remainder of the vehicle trajectory.

Prior to SA-2's flight, laboratory
experiments in connection with "Project
High Water" were conducted at Marshall
Center's Astrionics Division.

Saturn flight conditions are simulated
by using a vacuum chamber. To facilitate
viewing, coloring is added to the water
in the test tube.

In this experiment, the tube is suspended
in a horizontal position. A solenoid-
operated hammer breaks the tip releasing
the water. Because of the low pressure,
the water evaporates rapidly. Cooling
is so fast that ice flakes form immediately.
With the tube in this position, water boil-
off is slow and sporadic.

FILM

Sequence no. 14:

slow motion -- CU of
chamber, hammer breaking
tip of vertical tube, etc.

Sequence no. 15:

Static test firing
of SA-3 (include shots
which show a lot of
engine gimbaling)
(0450 and 0453)

NARRATION

In a second experiment, a vial is suspended vertically. The tip is broken, and a boiling reaction occurs. With the vial vertical, water boil-off is constant. In both experiments, pressure is so low that the ice which is formed has an unusually low temperature is very hard and elastic.

(SOUND EFFECTS: SATURN BOOSTER STATIC FIRING. THEN, LOWER SOUND EFFECTS, AND VOICE OVER--)

Three static test firings of the third Saturn flight vehicle, SA-3, were held at Marshall this quarter, two of 30 seconds duration and the final one running 119 seconds. Defective bearings and mainshafts resulted in excessive turbo-pump vibration in the first test. Defective parts were replaced, and pumps and engines were satisfactorily re-tested before the engines were re-installed. Later firings encountered no difficulty.

FILM

Sequence no. 16:

Final assembly of SA-4

Sequence no. 17:

SA-5 thrust frame

barrel assembly;

men working on it

Sequence no. 18:

New SA-5 spider

beam fabrication

fixture; then show

completed spider beam

(0478)

Sequence no. 19:

Overall view of

explosive forming

tank and supporting

equipment

NARRATION

Assembly of the booster for the fourth Saturn flight test vehicle, SA-4, was completed on May 28th, and the stage is now undergoing pre-static-test checkout.

Fabrication of components and sub-assemblies, such as this thrust frame barrel assembly, for the fifth Saturn flight booster, SA-5--first of the Block II series--was carried out this quarter by Marshall's Manufacturing Engineering Division.

A number of new fabrication fixtures, such as this one for making Saturn spider beam assemblies, have been placed into service for Block II booster fabrication.

Looking toward future fabrication techniques for Saturn or other space vehicles, Marshall engineers are investigating exploding bridge-wires in a fluid media as a means of forming and working metals.

FILM

Sequence no. 20:

men loading metal
stock onto die

Sequence no. 21:

Crane lifting die
into tank

Sequence no. 22:

Explosion in tank

Sequence no. 23:

Men removing die
from tank; men
removing cover to
reveal formed metal

NARRATION

In this test, a piece of flat stock
aluminum is loaded onto a female die
and securely mounted.

A crane hoists the die and stock into
the forming tank, which is filled with
water, and the exploding bridge-wire
is properly positioned.

(SOUND EFFECTS: INSERT CLAP-STICK SOUND
AT APPROPRIATE TIME IN NARRATION--BUT NO
SO LOUD AS TO DROWN OUT THE NARRATOR.)

The ultra-fast discharge of a large
capacitor bank explodes the bridge-wire,
creating a high-energy shock wave in
the water.

This shock wave, along with hydrodynamic
pressure pulses, forms the metal into the
previously evacuated die. Advantages of
forming materials by this method lie in
the control of forming and relative ease
of operation.

FILM

Sequence no. 24:

Establishing shorts of

Hayes International

(sign)

(0489 and 0477)

(also, maybe 0494)

Sequence no. 25:

Overall view of fins

mounted in jig fixture

(0477); LS of large fin

(0477)

Sequence no. 26:

LS of stub fins;

CU of drawings of fins;

then back to actual fin

(0477)

NARRATION

Hayes International, Inc. in Birmingham, is fabricating several Block II booster components, including fins, lower shrouds, and engine skirts.

Fin design utilizes the spar, rib, and skin type structure, which provides a high degree of structural reliability. Three basic fin configurations are used. Four large fins will be located at 90-degree intervals around the booster.

Two configurations of stub fins will be located at right angles to each other between the large fins. Three of these have provisions for venting liquid hydrogen from the vehicle's second stage. In addition to providing flight stability, these eight fins have vehicle support and hold-down fittings.

FILM

Sequence no. 27:

Lower shroud

(0477)

Sequence no. 28:

Establishing shots
of Republic Aviation
(aerial and ground)

Sequence no. 29:

Saturn manufacturing (tape-
controlled machines,
use of tape, tape being
loaded, milling machinery,
welding, finished parts
being carted away.

(Film may run a bit longer
than narration.)

Sequence no. 30:

Lox tank being moved
into MSFC shop by
rail car
(0458 and maybe 0456)

NARRATION

The lower shroud which Hayes makes for Saturn Block II boosters is basically a corrugated skin structure with continuous rings supporting the entire unit.

Republic Aviation Corporation, of Long Island, New York, is another prime example of industry at work for Saturn.

One of the world's largest banks of numerical control machines--which operate from taped manufacturing instructions--is being put to use by Republic Aviation for fabrication of a large number of various Saturn components.

The first of the Saturn Lox and fuel tanks manufactured by Ling-Tempco-Vought, near Dallas, Texas, were delivered to the Marshall Center this quarter.

FILM

Sequence no. 31:

LS of shipping container;
tank being removed from
container

Sequence no. 32:

Men inspecting tank;
man using leak probe
in checking pressure
leaks in tank

Sequence no. 33:

Stock footage showing
row of several H-1
engines in ME Division

Sequence no. 34:

Man machining inner
chamber of model H-1
engine on lathe; welding;
drilling; assembly of
completed model

NARRATION

During transportation, the tanks are
protected from damage by a custom-made
shipping container.

Marshall personnel thoroughly inspect
each tank prior to acceptance. Tanks
are subjected to an air pressure leak test,
with freon used as a tracer gas. If
leakage exists, an electronic instrument
detects the area of escaping gas.

Delivery of the H-1 engines, both inboard
and outboard, for the SA-5 booster was
accomplished early in April by the
contractor, Rocketdyne Division of
North American Aviation Company.

Small model rocket engines, such as this
500-pound-thrust H-1 model, are being
fabricated by the Marshall Center's Test
Division for use in gathering data about
their real counterparts.

FILM

Sequence no. 35:

Overall exterior view
of high-altitude test
chamber at Test Div.,
used for S-1/S-IV
interstage separation
studies

(0454)

Sequence no. 36:

Interstage separation
testing (engineering
footage shot from
inside chamber)

(0454)

Sequence no. 37:

Men installing model
of Block II booster
and model of deflector
plate

NARRATION

One-tenth scale models of the C-1 Saturn's
booster and S-IV stage have been tested
in a high-altitude chamber to study
interstage separation problems.

(SOUND EFFECTS: MODEL ENGINE STATIC FIRING.
VOICE OVER--)

Test objectives were to obtain data on
pressure versus interstage separation
distances, and to determine the effect
of a modified conical flow deflector on
the hot gas backlash.

A 1/20th-scale model of a Block II Saturn
booster was tested in conjunction with
a model flame deflector of the type intended
for use on the launch pedestal of Launch
Complex 37, now under construction at
Cape Canaveral.

FILM

Sequence no. 38:
Static firing of
Block II booster model

Sequence no. 39:
New Block II booster
assembly station
(0476)

Sequence no. 40:
Men working on
SA-5 tooling ring

Sequence no. 41:
Establishing shots
of IBM plant, Owego, N. Y.

NARRATION

(SOUND EFFECTS: MODEL SATURN BOOSTER
FIRING. THEN, VOICE OVER--)

This test program enables engineers to study base region environmental pressures, temperatures, and heating rates, as well as flame deflector effectiveness under hot-firing conditions.

The new Block II Saturn booster assembly station was installed during this report period in Marshall's recently expanded Saturn assembly building, which now contains over 200,000 square feet of floor space.

The tooling ring for the SA-5 booster has been fabricated, and work is scheduled to begin in July on SA-5 booster assembly.

Selection of International Business Machines, Inc., Owego, New York, to develop the guidance computer for Saturn C-1 was announced this quarter. For test purposes, the computer will be aboard the SA-5.

FILM

Sequence no. 42:

no men examining
camera pod; Section
of spider beam; camera
pod and recovery package being
mounted into ejection
cylinder; umbilical
connection made

Sequence no. 43:

Sphere pressure

Sequence no. 44:

Ejection of camera
pod and recovery
package

NARRATION

Also slated for initial use on SA-5 is
a new camera-eject mechanism which will
help to provide a photographic record of
vehicle actions. Along the spider beam of
the SA-5 booster, eight movie camera pods
and para-balloon recovery packages will be
mounted into ejection cylinders.

In this laboratory test at the Marshall
Center, gaseous nitrogen is used as a
pressurant.

When sufficient pressure is attained,
the firing switch is closed and the camera
pod and recovery package are ejected.

(SOUND EFFECTS: CLAP-STICK.)

FILM

Sequence no. 45:

removal of SA-D
upper stages and
booster from dynamic
test stand

Sequence no. 46:

Removal of old Jupiter
test position from MSFC
static test stand (if
footage available; if
not, scene showing stand
after removal)

Sequence no. 47:

Artist's drawing of
modified Saturn
static test stand

NARRATION

SA-D, the test vehicle which had provided vital dynamic vibration data contributing to the success of the first two flight vehicles, was removed from Marshall's dynamic test stand this quarter, its mission completed. A new vehicle, SA-D-5, a simulation of SA-5, will be built at Marshall and later installed in the stand for testing.

Marshall's static test stand will soon be modified to accommodate two Saturn C-1 boosters simultaneously. The old test position in which Jupiter and Juno II rockets were once tested has already been removed in preparation for creating...

... a second Saturn booster test position in its place.

FILM

Sequence no. 48:

Construction of
new MSFC headquarters
(steel framework)
(0491)

Sequence no. 49:

New five-story
P&VE building

Sequence no. 50:

New ME Division building

Sequence no. 51:

ME Division and
Chrysler personnel
at table (Chrysler
people receiving
orientation on Saturn)
(0483)

NARRATION

Several major construction projects are changing the Marshall Center horizon. The nine-story Central Laboratory and Office Building, scheduled for completion next January, will be the Center's tallest building.

Personnel of the Propulsion and Vehicle Engineering Division are due to begin occupying their new five-story addition in July.

And Manufacturing Engineering Division has already moved into its recently-finished addition.

At ME Division, a group of Chrysler engineers and technicians are presently receiving orientation on Saturn fabrication and assembly methods, in preparation for Chrysler's future C-1 booster manufacturing at Marshall's Michoud Operations plant near New Orleans.

FILM

Sequence no. 52:

Establishing shots
of new NASA
building at Slidell, La.
(0480)

Sequence no. 53:

Artist's drawings
of Mississippi
Test Facility
(479 and 0482)

Sequence no. 54:

Overall aerial view
of VLF 37
(PL 62-66731)

Sequence no. 55:

Service tower

NARRATION

Twenty miles from Michoud, at Slidell, Louisiana, this new two-million-dollar building has been acquired by NASA from the Federal Aviation Agency. The building, which contains 53,000 square feet of floor space, is being occupied by some 500 Chrysler employees, in a move to alleviate a critical office space problem at Michoud.

At the Mississippi Test Facility site, negotiations are now underway with some 200 land-owners in the construction area. The government's schedule calls for outright acquisition of title to the area by July 31st.

Construction of Saturn Launch Complex 37 continued at Cape Canaveral during this report period.

Work includes construction of the mobile 3500-ton steel service structure....

FILM

NARRATION

Sequence no. 56:

Umbilical tower and
launch pedestal

...268-foot-high umbilical tower and steel
launch pedestal...,

Sequence no. 57:

Blockhouse

...circular concrete blockhouse...,

Sequence no. 58:

Lox and fuel storage
facilities, and
servicing facilities

...Lox and fuel storage facilities, and
servicing facilities.

Sequence no. 59:

Overall view of
launch complex

Construction of major items is about 60
percent complete, and progressing on
schedule.

Sequence no. 60:

Artist's drawing
of VLF 37
(0460)

When finished, Complex 37 will have two
Saturn launch positions, utilizing a single
control center and service tower.

FILM

Sequence no. 61:

Choose from Scenes

, 3, 4, Douglas

Input no. 16

(Sacramento Test

Stand no. 1)

Sequence no. 62:

Scenes 18 and 19,

Douglas Input no. 21

(RL-10 engine)

Sequence no. 63:

Choose from Scenes

8, 9, 10, Douglas

Input no. 21

(Test Stand No. 2)

Sequence no. 64:

Choose from Scenes

11--16, Douglas Input

no. 21

(Modification of Test

and no. 21)

NARRATION

At Douglas Aircraft Company, contractor for Saturn's S-IV stage, cold flow tests have been successfully completed at the Sacramento test facility, using a single RL-10 liquid-hydrogen, liquid-oxygen engine.

Five additional engines were received this quarter from Pratt and Whitney. After acceptance checking at Santa Monica, the engines were shipped to Sacramento and installed in the battleship test vehicle in preparation for the second phase of the battleship test program.

Modification of Test Stand No. 2, which will be used for the All-systems testing, continues on schedule.

The steam system was being installed during this report period, and other necessary hardware is now available for completion of the stand.

FILM

NARRATION

Sequence no. 65:

(NOTE: Footage in this section is from Cornell Aeronautical Laboratory film entitled "Base Heating," plus additional footage of six-engine S-IV model supplied by Cornell.)

Establishing shots of
Cornell Aeronautical
Laboratory; animation
of hand drawing sketch
of base heating gases

The Cornell Aeronautical Laboratory, Buffalo, New York, has been conducting a series of tests with an S-IV model in an altitude chamber, looking toward solution of problems which occur when a portion of the engines' hot exhaust gas escapes from the exhaust plume and flows into the base region.

Sequence no. 66:

(NOTE: All the following sequences appear in same order as they do in the Cornell film, "Base Heating.")

During this test, which lasts for only five-thousandths of a second, pressure and temperature measurements are taken on the base plate of the model, using miniature, highly sensitive instruments. Piezo-electric (NOTE TO NARRATOR: PRONOUNCE IT "PIE-EE-ZO") pressure transducers are mounted behind orifices in the base plate at locations where pressure is to be read.

FILM

Continue action

NARRATION

Fragile thermometers consist of a thin film of metallic paint applied to a quartz button. When the surface of the button is heated by the gas, the electrical resistance of the metallic film changes. Then the output voltage signal of the thermometer denotes the instantaneous temperature of the particular location under survey.

Continue action

By observing the time history of this temperature, the local heating rate is determined.

Continue action

Fast-responding instruments such as these permit Cornell Aeronautical Laboratory Scientists to study rocket base heating problems in short-duration experiments.

Continue action

Such tests are better controlled and much more economical to perform than conventional techniques involving continuous operations.

FILM

Continue action

Sequence no. 67:

Choose from Scenes

2506-2447, P&W April

input

(Preparation for firing)

Sequence no. 68:

Choose from Scenes

2719-2773, P&W April

input

(static firing)

Sequence no. 69:

Choose from Scenes

683-698, P&W April input

(RL10A-3 engine gimbaling)

NARRATION

Here is one frame taken from a high-speed Schlieren motion picture film, showing shock waves created by the combusted gases exhausting into the vacuum chamber.

Preliminary Flight Rating endurance testing of the S-IV stage's RL10A-3 engine was successfully completed on June 9th by the engine contractor, Pratt and Whitney, at West Palm Beach, Florida.

(SOUND EFFECTS: RL10A-3 ENGINE STATIC FIRING. THEN, LOWER SOUND EFFECTS, AND VOICE OVER--) Twenty-six PFRT firings, totaling 4,096 seconds, were conducted. Initial inspection showed the engine to be in good condition.

A series of non-firing gimbal tests of the RL10A-3, using Douglas Aircraft Company plumbing connections, was also carried out.

FILM

Sequence no. 70:

Choose from Scenes

323-328, P&W April input

(stress coating engine)

Sequence no. 71:

Choose from Scenes

699-712, P&W April input

(engine gimbaling)

Sequence no. 72:

Choose from Scenes

779-788, P&W April input

(plumbing connections)

Sequence no. 73:

Choose from Scenes

1625-2647, P&W April input

(Vertical stand and LH2 tank)

NARRATION

To test engines and hardware for possible structural weakness, a stress coat was applied on metal surfaces to locate areas of structural yield.

Various gimbal angles and frequencies were applied to the engine to simulate the worst expected flight conditions.

Both engine and vehicle plumbing withstood the tests satisfactorily.

In support of the engine program, facilities completed at Pratt and Whitney's Research and Development Center this quarter included a new vertical single engine test stand...and a 90,000-gallon vacuum-jacketed liquid hydrogen spherical storage container.

FILM

Sequence no. 74:

Artist's drawing,
showing C-1 Saturn
alongside Statue of Liberty

Sequence no. 75:

PULL BACK To show
C-5 Saturn alongside
C-1 and Statue

Sequence no. 76:

Model of Saturn C-5

Sequence no. 77:

Latest construction
shots at new static
test stand site

(0490)

NARRATION

As progress continued this quarter on the Saturn C-1, shown alongside the Statue of Liberty in an artist's conception to dramatize its great size...,

...work was also underway on the even larger Advanced, or C-5, version of Saturn. The C-5 will stand about 350 feet tall, as compared to 170 for C-1.

The C-5, shown in model form, will be able to hurl over 200,000 pounds into a 300-mile orbit. The vehicle could use two stages for earth orbit missions and three stages for escape missions. Launching of the first C-5 is expected in 1965.

At Marshall, construction is proceeding on the static test facility to be used for testing C-5 boosters. The concrete foundation for the massive stand plunges over 45 feet into the earth. Including its crane, the new test structure will be 405 feet tall.

FILM

Sequence no. 78:

Boeing sign and
entrance at HIC Building;
employees entering; interior
of building
(0494)

Sequence no. 79:

Establishing shots
of North American
Aviation headquarters
and plant

Sequence no. 80:

Scene 5,
S-II April input
(model engines and
flame deflector)

NARRATION

Over 1,000 employees of the Boeing Company, contractor for the Saturn C-5 booster, are now at work in the Huntsville area. The company is expected to employ more than 1,500 there during 1962, most of whom will later be transferred to Marshall's Michoud Operations where the giant boosters will be manufactured.

At North American Aviation's Space and information Systems Division, contractor for the Saturn C-5's S-II, or second stage, work this quarter included...

..."hot-flow" tests using scale model engines, with a model flame deflector of comparable scale, to determine optimum engine orientation for the five-engine S-II configuration.

FILM

Sequence no. 81:

S-II April input
(static firing of
model engine)

Sequence no. 82:

Scene 7,
S-II April input
(Model engine and
deflector after firing)

Sequence no. 83:

Scenes 8, 9, 10,
S-II April input
(model of S-II stage)

NARRATION

(SOUND EFFECTS: MODEL ENGINE BEING
STATIC FIRED. THEN, LOWER SOUND EFFECTS,
AND VOICE OVER--)

Secondary objectives of the tests include
determination of various deflector parameters,
such as pressure, temperature, and heat
flux profiles, plus investigations of film
coolant injection methods.

The scale model engines produce a total
thrust of 5,000 pounds. The deflector
is coated with zinc chromate paint, which
burns away during firings to reveal areas
of probable burn-through.

Fabrication of an S-II stage and transporter
model, designed to verify that the booster
transporter will meet all maneuverability
requirements, is now complete.

FILM

Sequence no. 84:

Choose from

scenes 24-32,

S-II April input

(Road gage survey)

NARRATION

Using a road gage fabricated to the same dimensions as the S-II transporter, a month-long survey has been conducted to determine the feasibility of routes proposed for overland transportation of the stage from Port Hueneme, (NOTE TO NARRATOR: PRONOUNCE IT "WAN--EE--ME") California, to North American's static test facility at Santa Susana, a distance of some 50 miles.

Sequence no. 85:

Scenes 12, 13, 14,

S-II April input

(Making plaster model
for lay-up die)

A plaster model has been made to serve as a tooling aid for constructing a female lay-up die, which will be used to form bulkhead gore segments for the S-II mockup. The sweeping frame employed in this operation will later be used to "sweep" the production tooling master.

FILM

Sequence no. 86:

Scenes 15, 16, 17, 18,

C-II April input

(C-5 antenna radiation
pattern model)

NARRATION

Two antenna radiation pattern models of the C-5 Saturn have been completed, and one has been shipped to the Los Angeles Division, where initial testing will be performed until the S&ID antenna range is operational. The program will determine the numbers and types of antennas required for telemetry, command control, and tracking aids, and will establish specific locations and angular orientation of antenna types selected.

Sequence no:87:

F-1 engine being

installed in

static test stand

(P-487)

A highly significant advance in the Saturn program occurred this quarter when the mammoth F-1 engine--five of which will be clustered for the C-5 booster--underwent its first full-duration static test at full thrust of 1.5 million pounds. The test was conducted at the NASA High Thrust Area at Edwards, California, by the F-1's developer, Rocketdyne Division of North American Aviation.

FILM

Sequence no. 88:

Static firing of

F-1 engine

NARRATION

(SOUND EFFECTS: F-1 ENGINE STATIC FIRING.
THEN, LOWER SOUND EFFECTS, AND VOICE
OVER--)

The ground test was sustained for 151.8
seconds, approximately flight duration,
before being terminated as programmed.

It was the longest test to date by the
only single rocket engine known to have
been operated above a million pounds of
thrust.

END